

LED-based Absorption Sensors for Early Fire and Hazardous Gases Detection for Flight Vehicles and Propulsion Engines

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Lillestrøm, Norway
June 11, 2018**



Organization

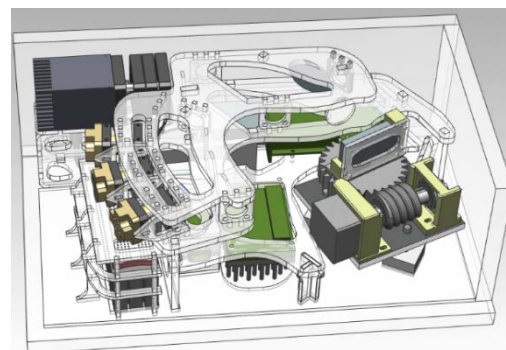
- ❖ Fundamentals of spectroscopy and absorption technique
- ❖ LED sensor design and lab validation
- ❖ Demo fire sensor flight test results





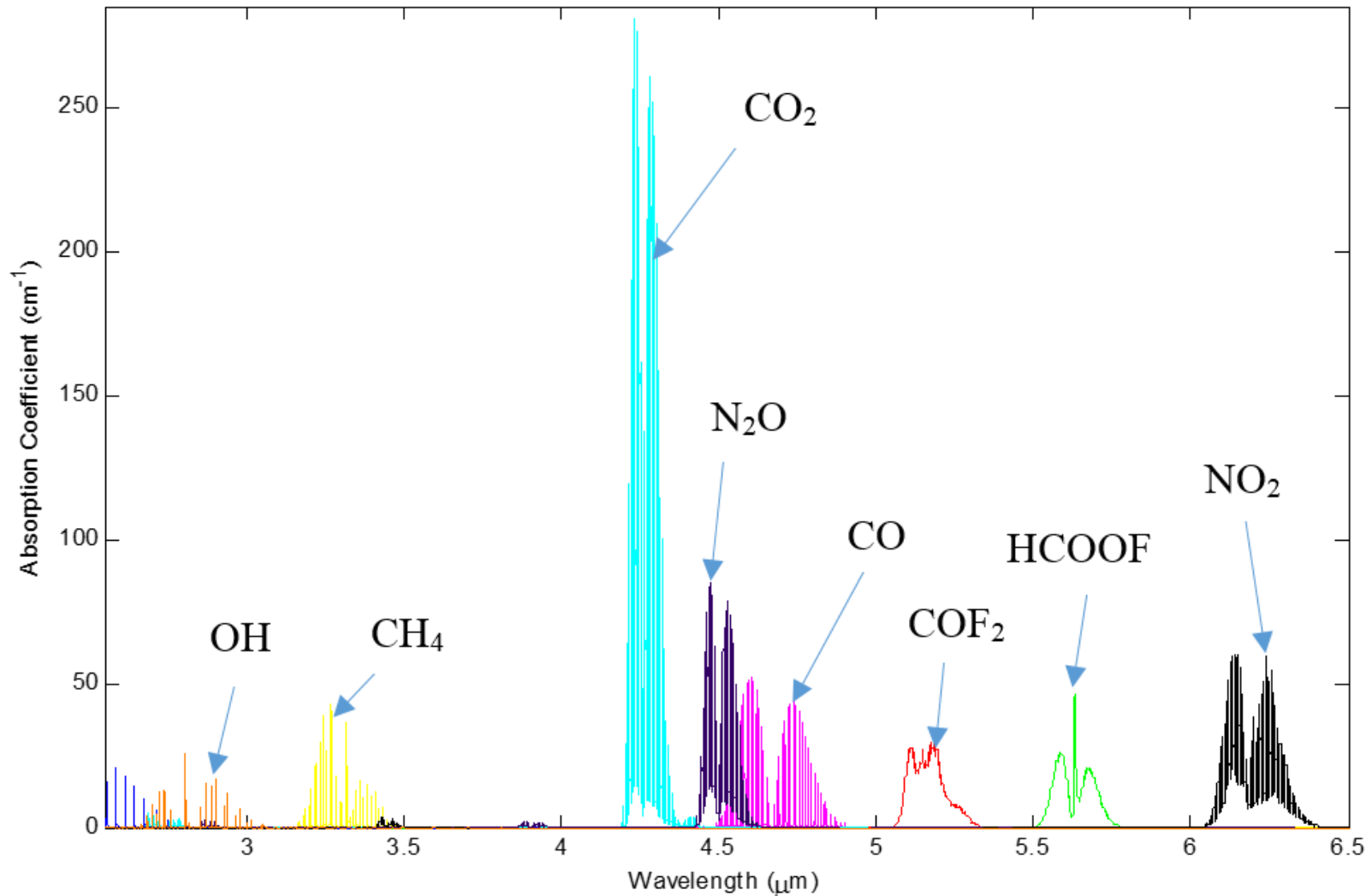
Organization

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Mid-Infrared Absorption Spectra





Absorption Spectroscopy and Beer's Law

Beer Lambert law of absorption

$$A_{\lambda} = -\ln\left(\frac{I_{\lambda}}{I_{\lambda,0}}\right) = k_{\lambda}L\chi$$

A_{λ} = spectral absorbance

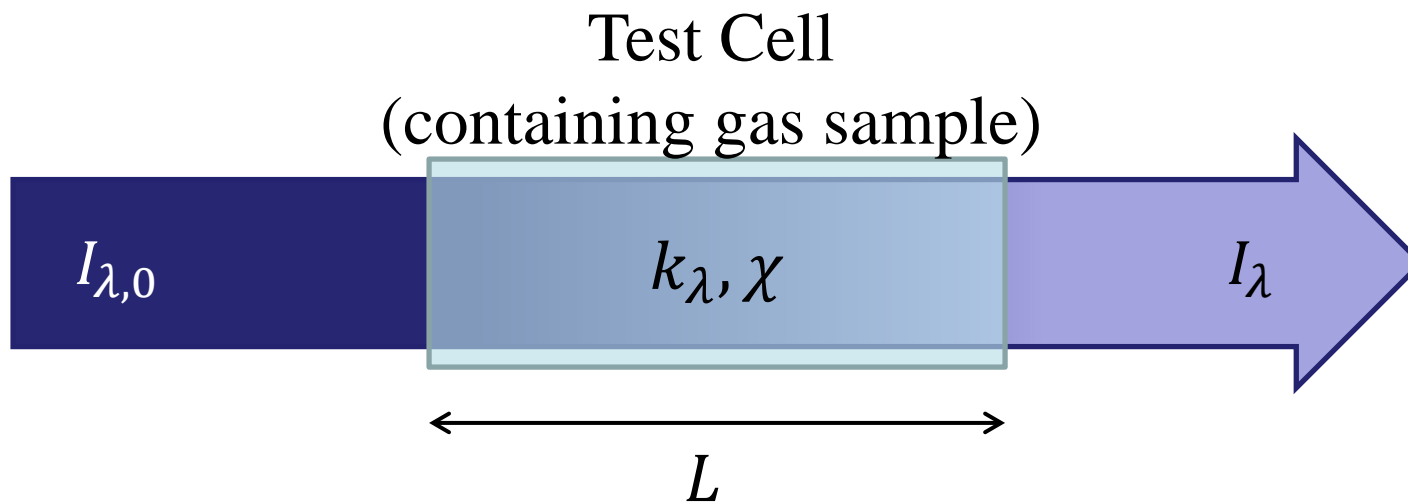
I_{λ} = transmitted radiation at λ

$I_{\lambda,0}$ = incident radiation at λ

k_{λ} = spectral absorption coef.

L = path length

χ = mole fraction of target gas





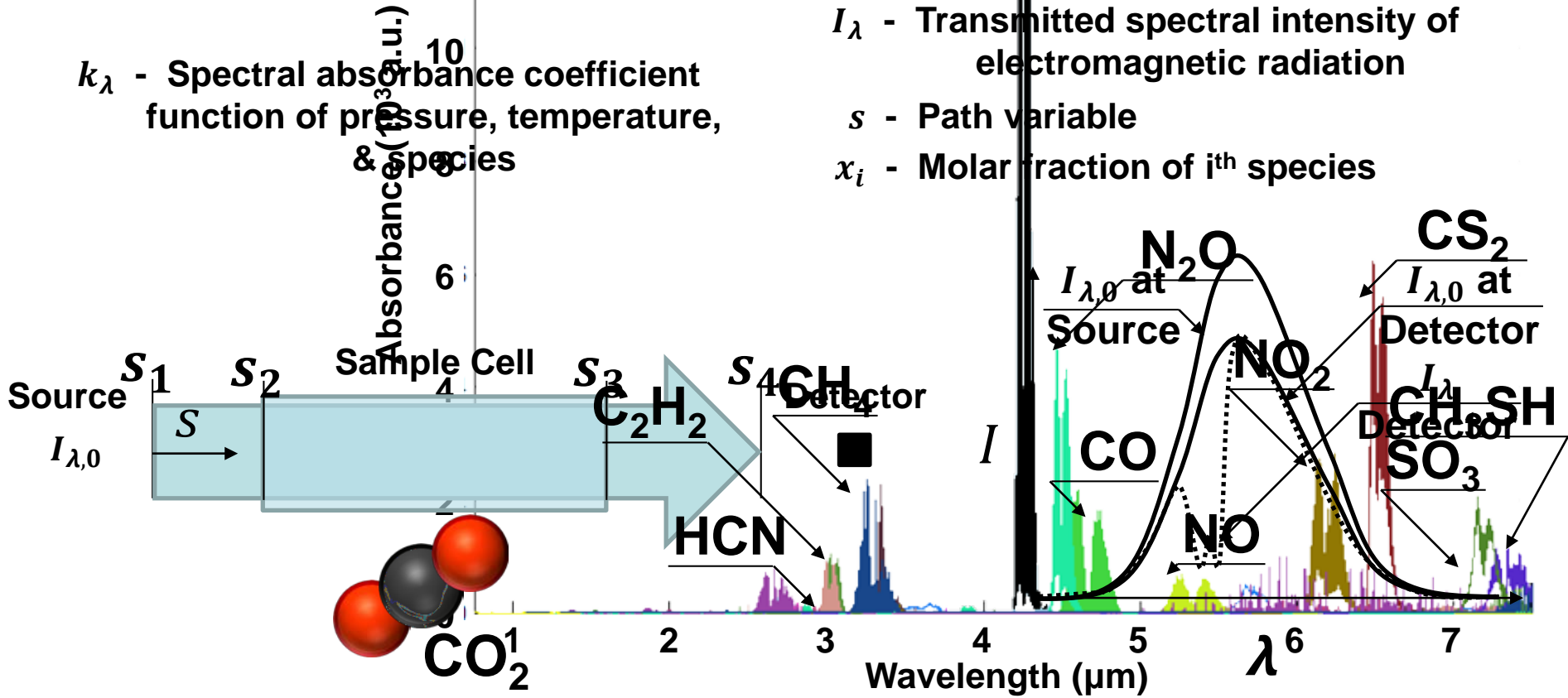
How Does Absorption Spectroscopy Work?

❖ Absorption Spectroscopy is an application of the Beer-Lambert Law

$$A(\lambda) = \ln\left(\frac{I_{\lambda,0}}{I_\lambda}\right) = \sum_i \int_{s_j}^{s_{j+1}} k_\lambda \cdot x_i ds$$

- $A(\lambda)$ - Spectral absorbance
- $I_{\lambda,0}$ - Incident spectral intensity of electromagnetic radiation
- I_λ - Transmitted spectral intensity of electromagnetic radiation
- s - Path variable
- x_i - Molar fraction of i^{th} species

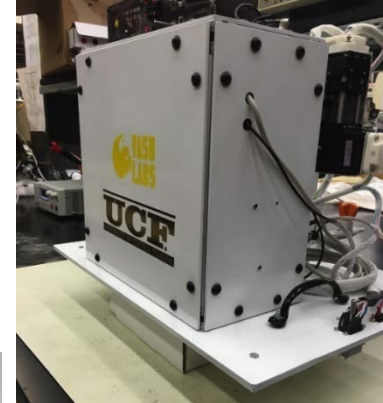
k_λ - Spectral absorbance coefficient function of pressure, temperature, & species





Organization

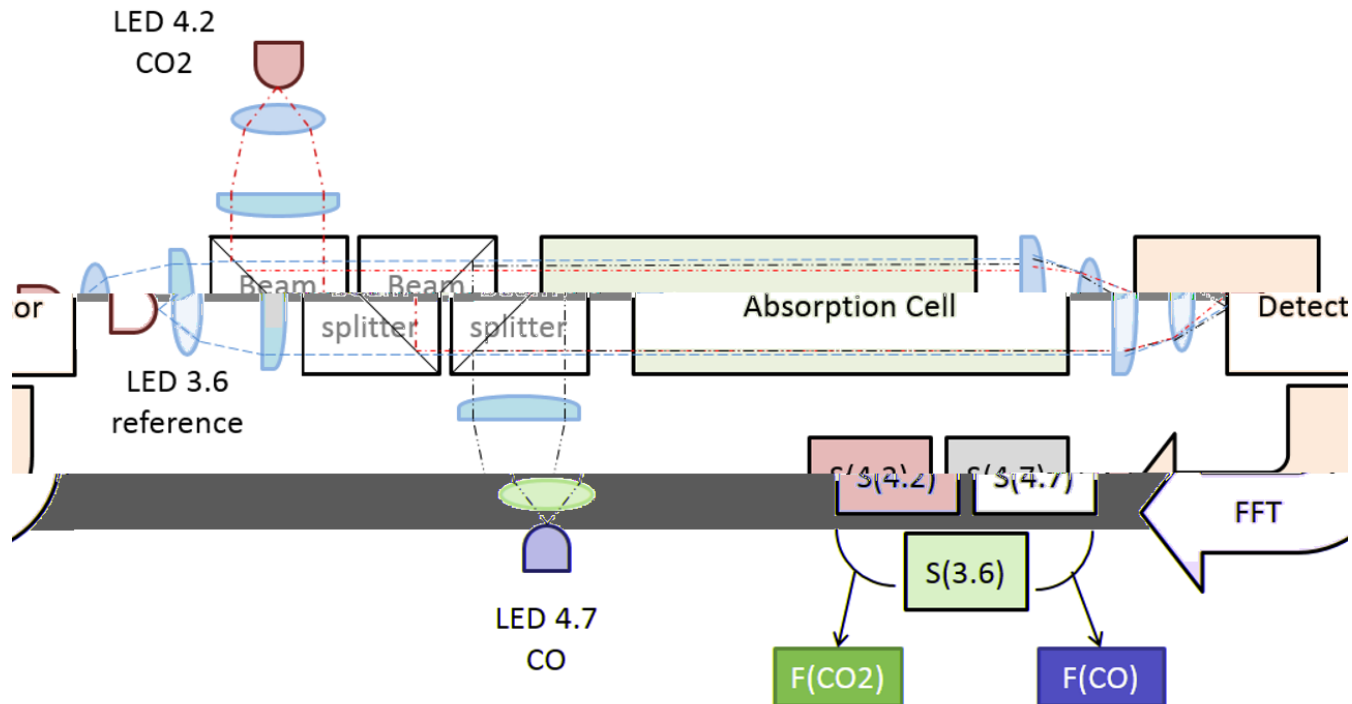
- ❖ Fundamentals of spectroscopy and absorption technique
- ❖ **LED sensor design and lab validation**
- ❖ Demo sensor flight test results





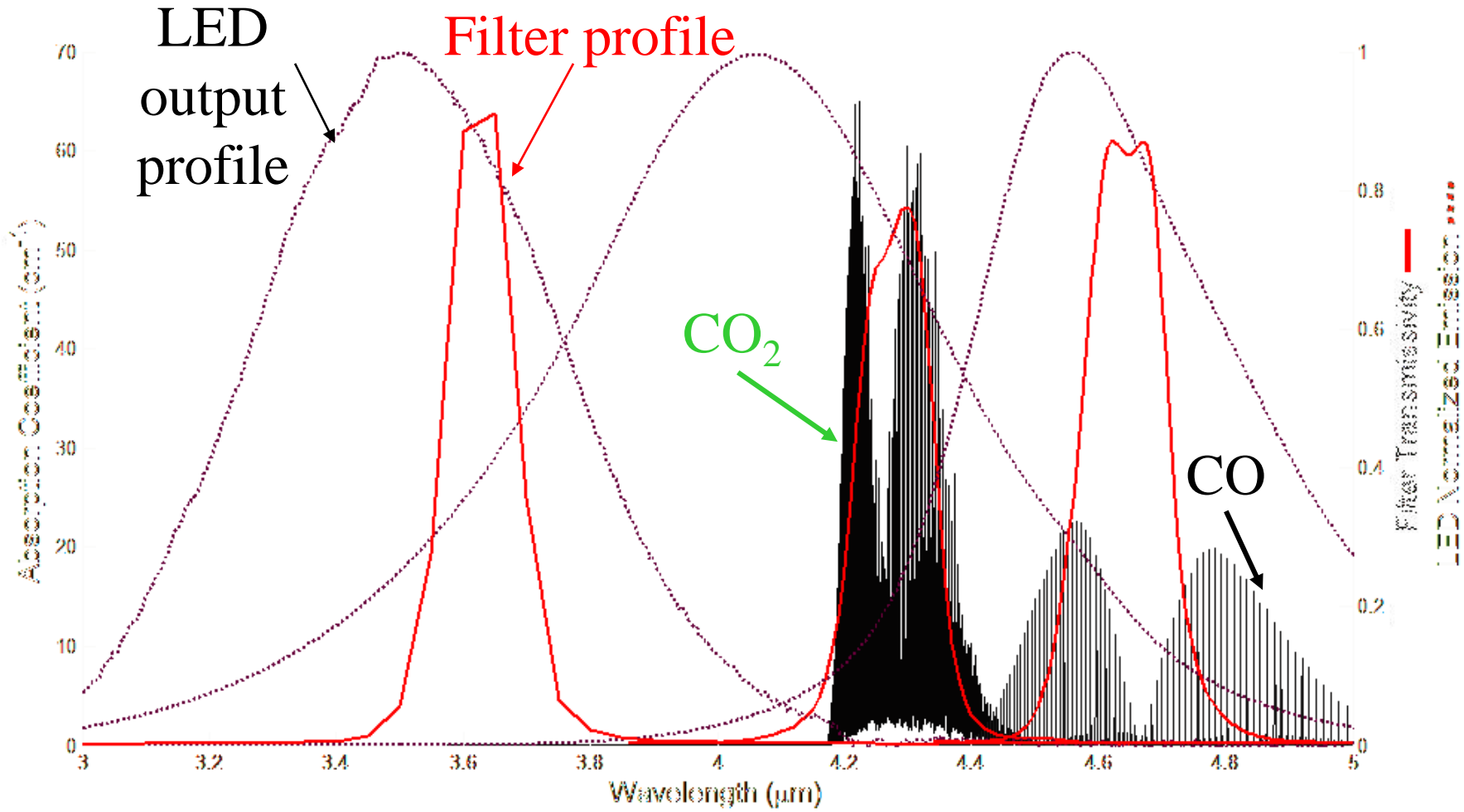
Sensor Design Using LEDs

- Three MIR LEDs centered at
 - 3.6 μm (for reference)
 - 4.2 μm (CO_2)
 - 4.7 μm (CO)
- LEDs amplitude modulated at different frequencies
- Band pass filters
- Collimating lenses
- Pellicle beam splitters
- Thermo-electrically cooled photovoltaic detector

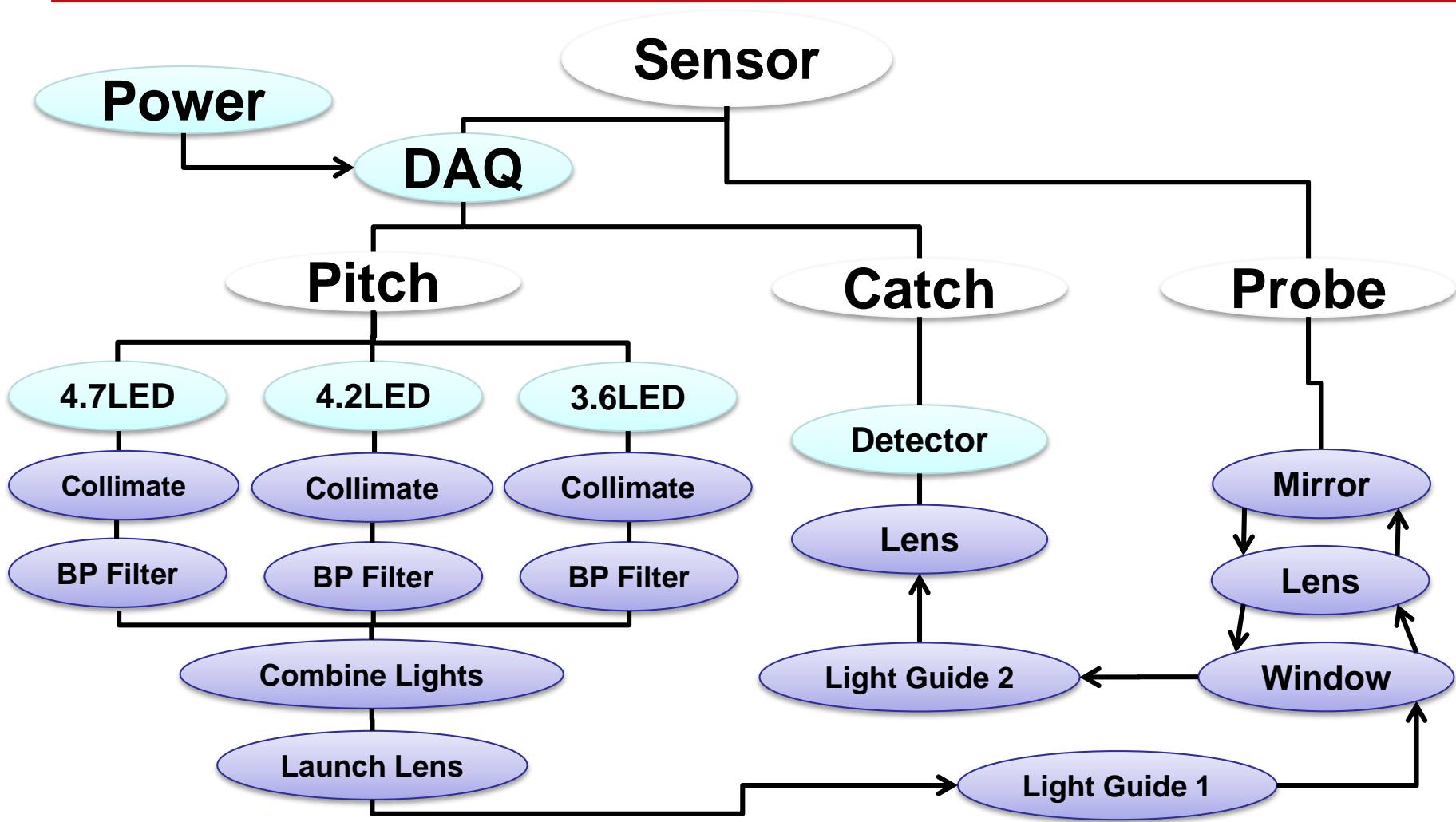




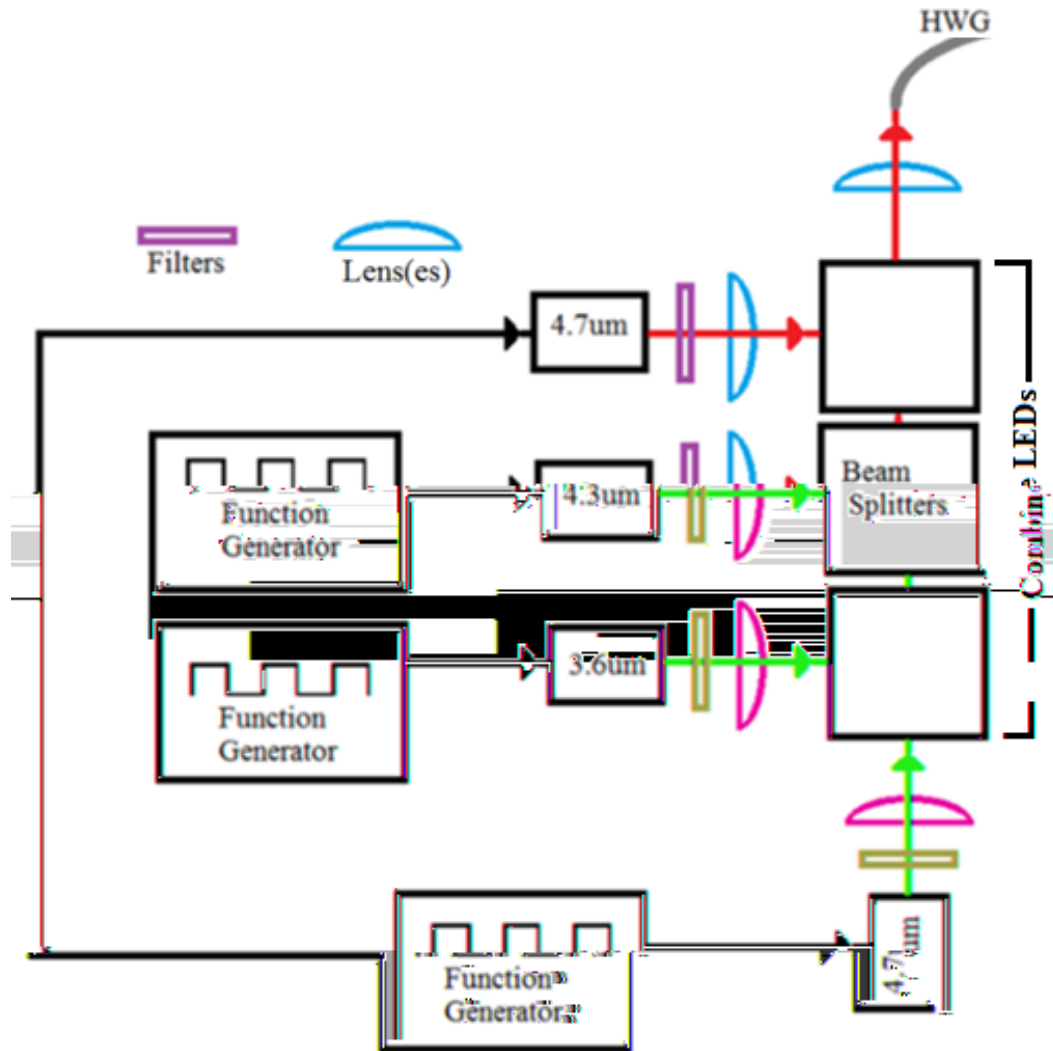
Broad spectrum of LEDs and absorption features of targeted gases



Sensor Overview and Operation



Pitch





Simulations using Zemax (Optic Studio)

Zemax 13 Premium - 23003 - C:\Users\Kyle\Dropbox\School\UCF\Research\Vasu\Design\LED Spectroscopy\Simulation\radialC1_47_4.ZMX

File Editors System Analysis Tools Reports Macros Extensions Window Help

New Ope Sav Sas Bac Res NCE MFE MCE TDE Upd Upa Gen Wav L3n LSn Obv Rtc Ltr Dvr Rdb Dis Gmp Opt Glb Ham Tol Gla ABg Sfv Xis Len Pre Chk Vop

Non-Sequential Component Editor

Edit Solves Tools View Help

Object Type	Comment	Ref Object	Inside Of	X Position	Y Position	Z Position	Tilt About X	Tilt About Y	Tilt About Z	Material
1 Source Ra..	LED Source	0	0	0.000	0.000	0.000	0.000	0.000	0.000	-
2 Standard ..	Col Lens 1	0	0	0.000	0.000	7.040	0.000	0.000	0.000	CAF2
3 Standard ..	Col Lens 2	0	0	0.000	0.000	57.400	V	0.000	0.000	CAF2
4 Standard ..	Launch Lens 1	0	0	0.000	0.000	199.900	P	0.000	0.000	CAF2
5 Standard ..	Launch Lens 2	0	0	0.000	0.000	222.000		0.000	0.000	CAF2
6 Cylinder ..	Pitch HWG	0	0	0.000	0.000	234.940		0.000	0.000	
7 Annulus		0	0	0.000	0.000	284.940	P	0.000	0.000	ABSORB
8 Standard ..	Cell col Lens 1	0	0	0.000	0.000	343.681		180.000	0.000	CAF2
9 Standard ..	Cell Col Lens 2	0	0	0.000	0.000	356.226		180.000	0.000	CAF2
10 Annulus		0	0	0.000	0.000	375.326	P	0.000	0.000	ABSORB
11 Annulus		0	0	0.000	0.000	455.032	P	0.000	0.000	ABSORB
12 Standard ..		0	0	0.000	0.000	474.226	P	0.000	0.000	CAF2

1: NSC 3D Layout

Update Settings Print Window Text Zoom

4: Detector Viewer 3

Update Settings Print Window Text Zoom

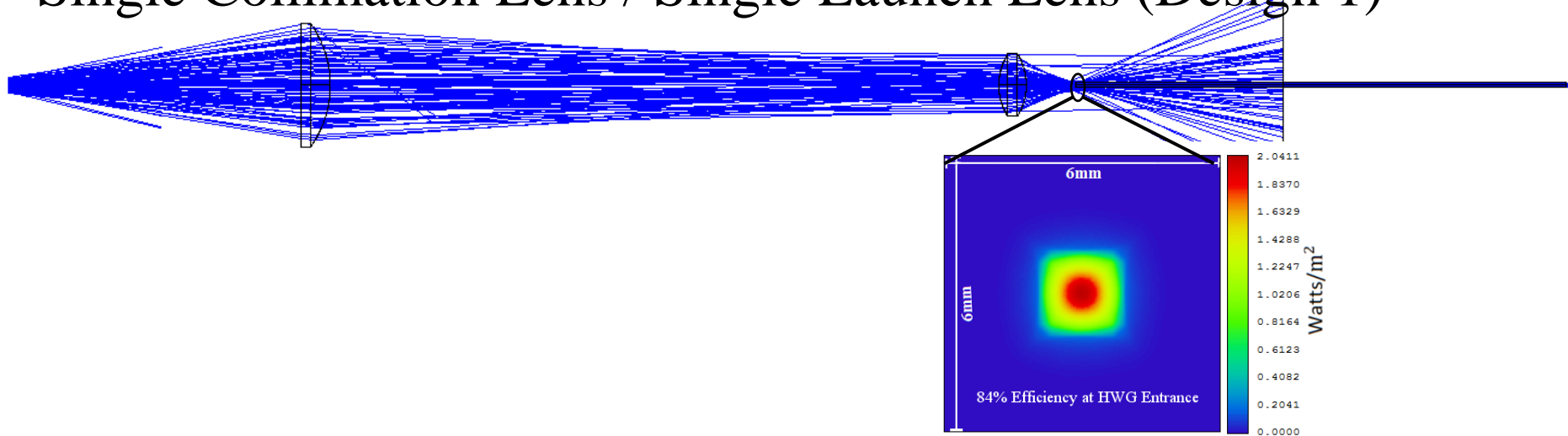
Detector Image: Incoherent Irradiance

10/6/2013
 Detector 19, NSCG Surface 1:
 Size 1.000 W X 1.000 H Millimeters, Pixels 500 W X 500 H, Total Hits = 2426178
 Peak Irradiance : 3.3897E+000 Watts/M^2
 Total Power : 1.2131E+000 Microwatts

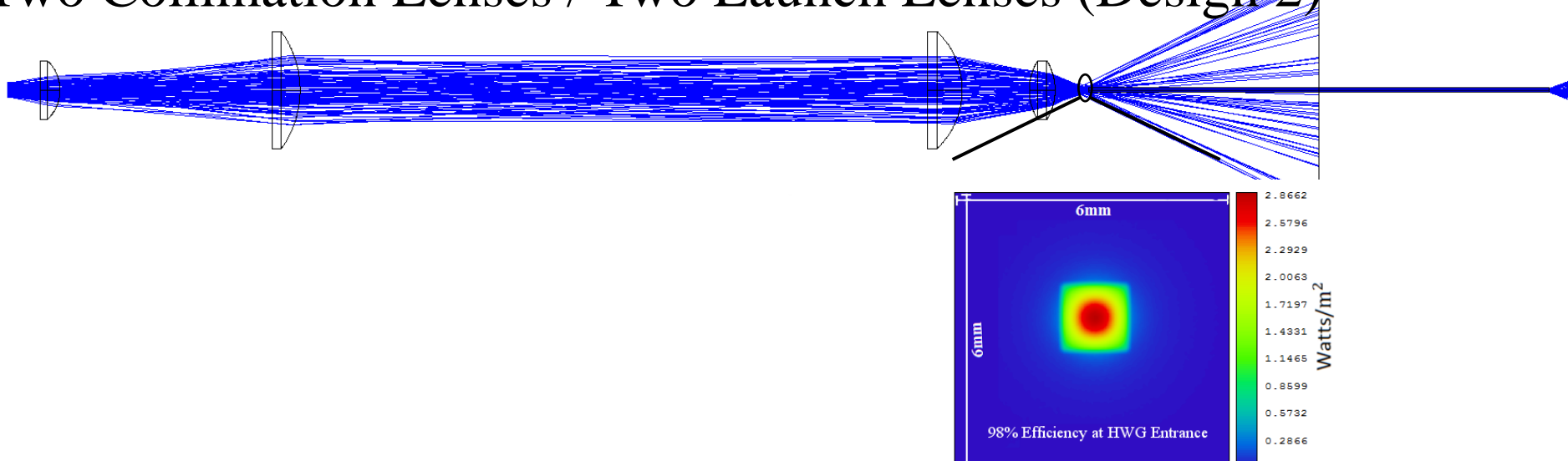


Simulations: Collimation

Single Collimation Lens / Single Launch Lens (Design 1)

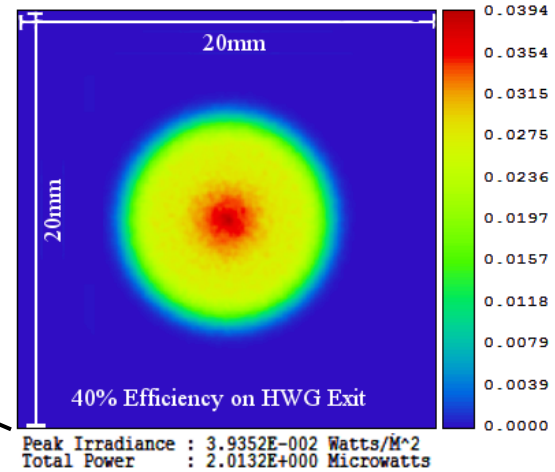
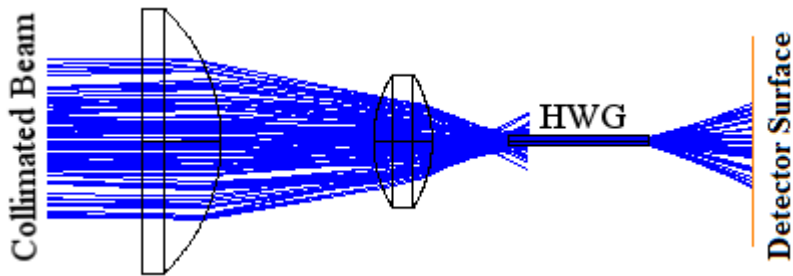


Two Collimation Lenses / Two Launch Lenses (Design 2)

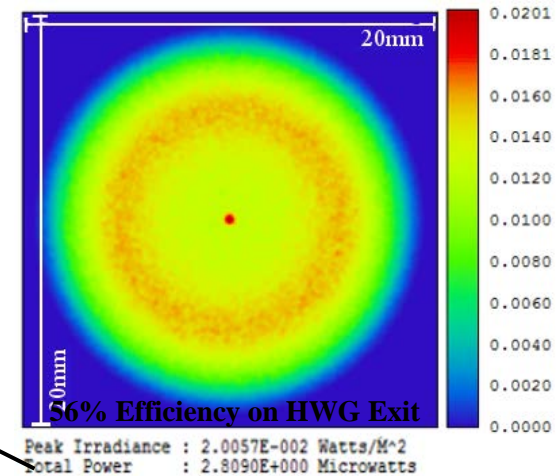
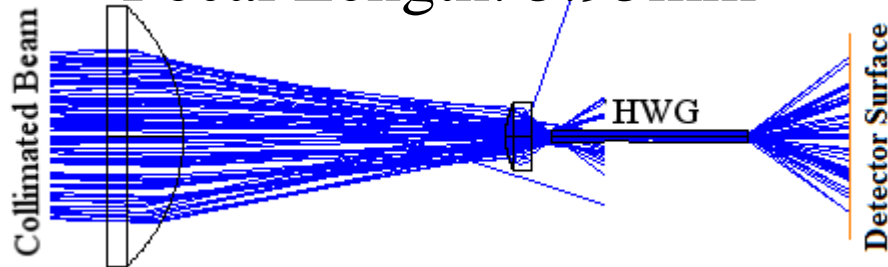


Simulations: Launch Lens FL

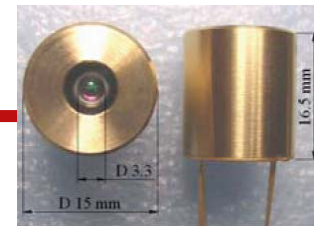
Focal Length: 15mm



Focal Length: 5.95mm



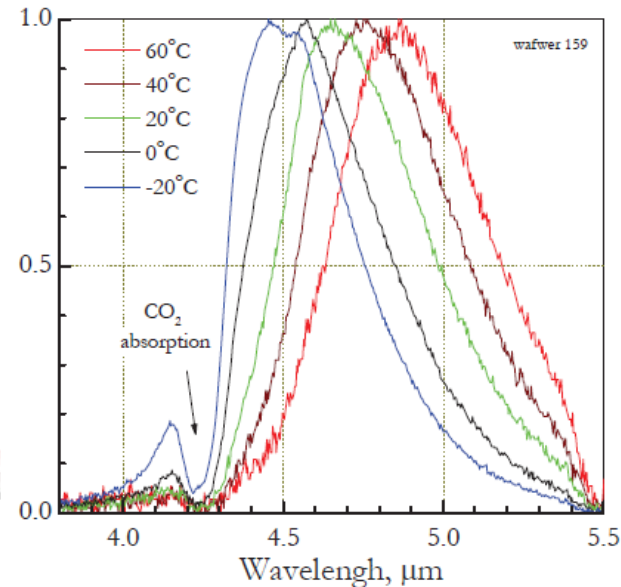
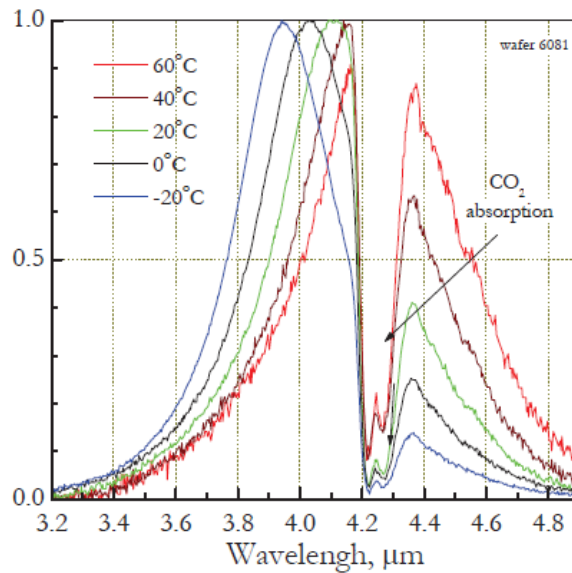
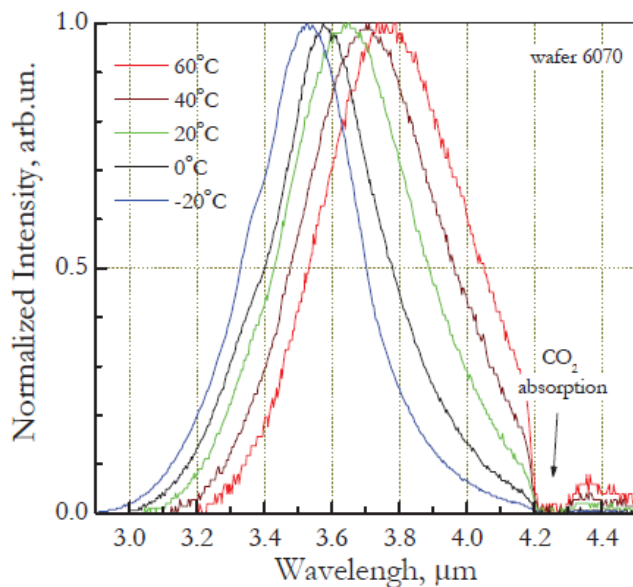
LED temperature control



3.6 μm LED w/TEC

4.2 μm LED w/TEC

4.7 μm LED w/TEC



Features

Far-Field Pattern

Axis Deviation

Emissions Size

All 3 LEDs

≤ 20 , deg

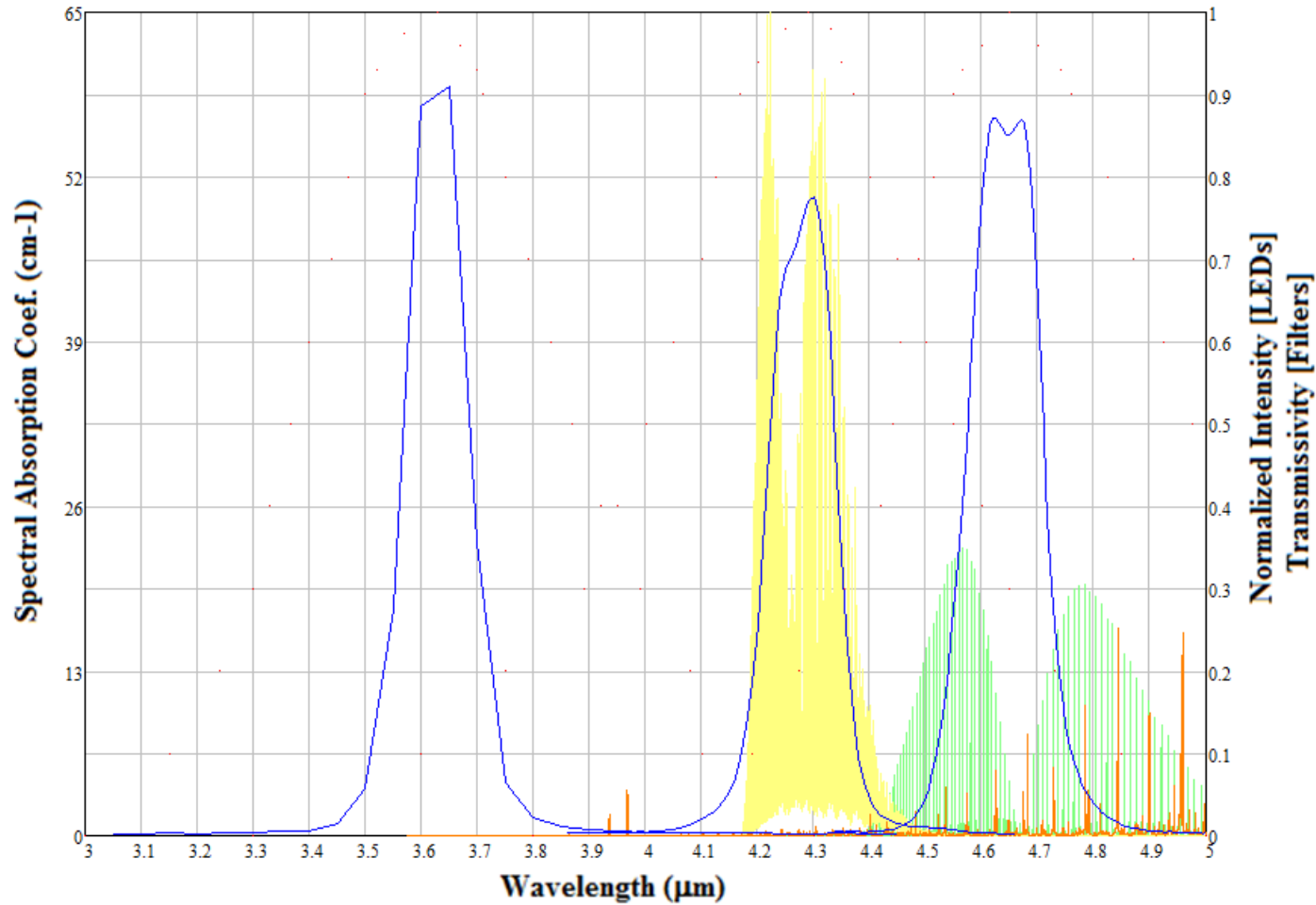
≤ 7 , deg

3.3mm (radius)

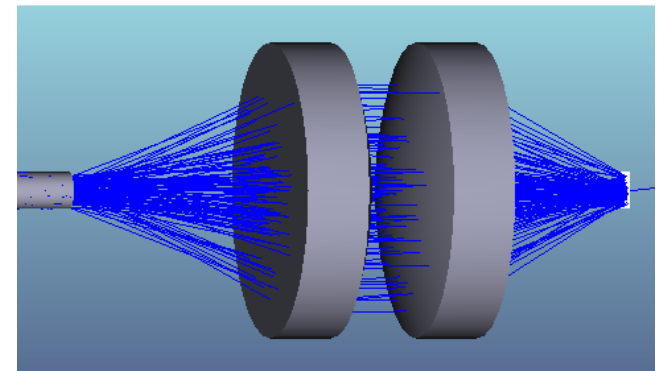
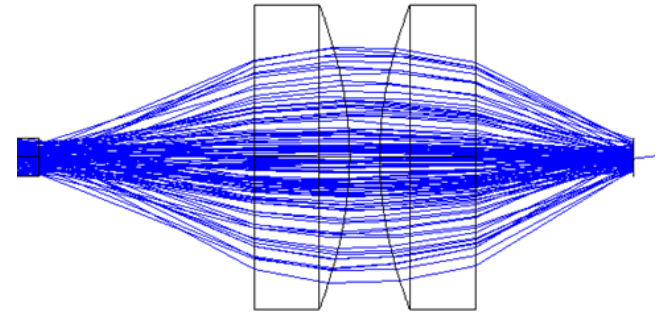
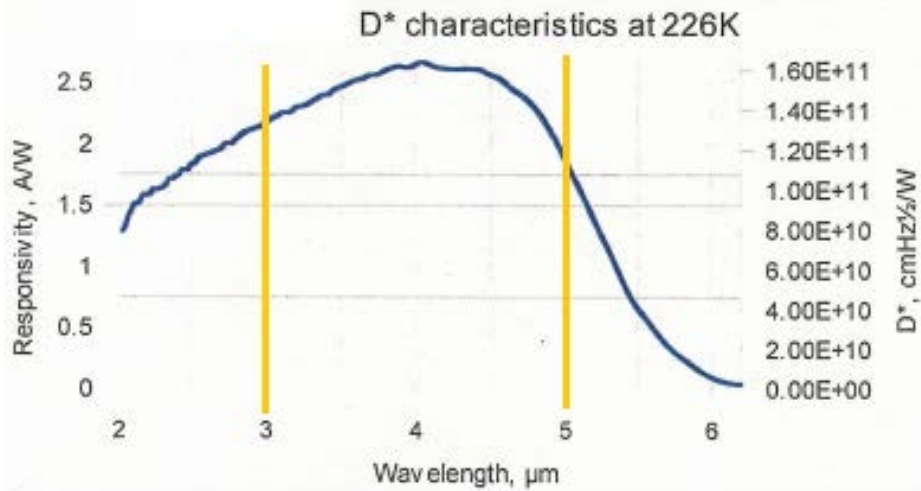


Filters Selection

- LED Output
- Filters
- CO (kv)
- CO2 (kv)
- H2O[x50] (kv)



Detector & Catch Optics

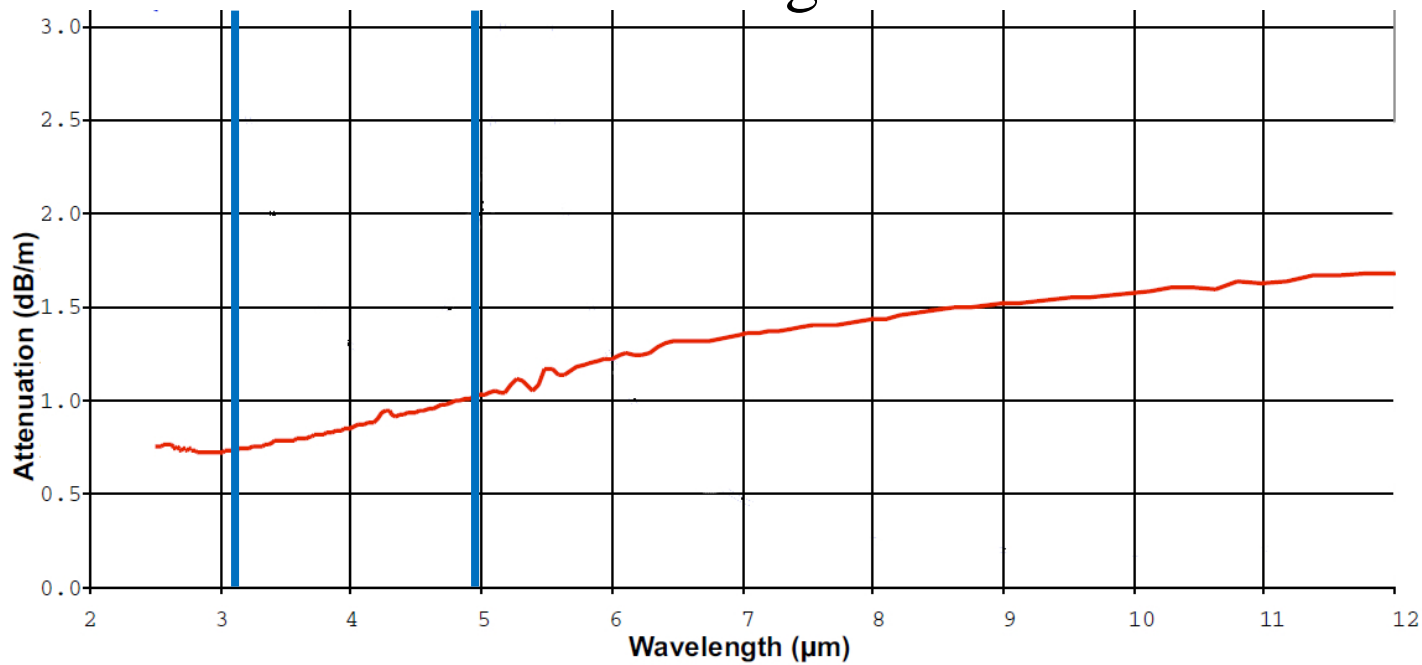


- Lenses: 2x BD-2 Aspheric Lens
 - Effective Focal Length: 5.95mm
- Detector: VIGO Systems PVI-2TE-5
 - Two stage, thermoelectrically cooled



Light Guiding

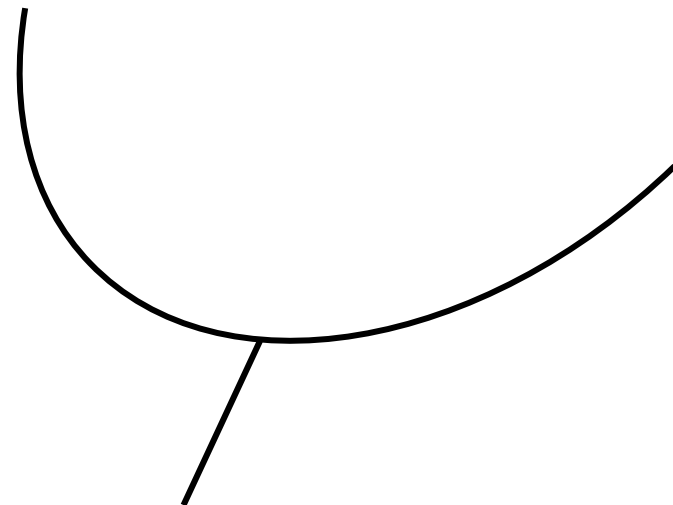
Hollow Silica Waveguide from Moxley



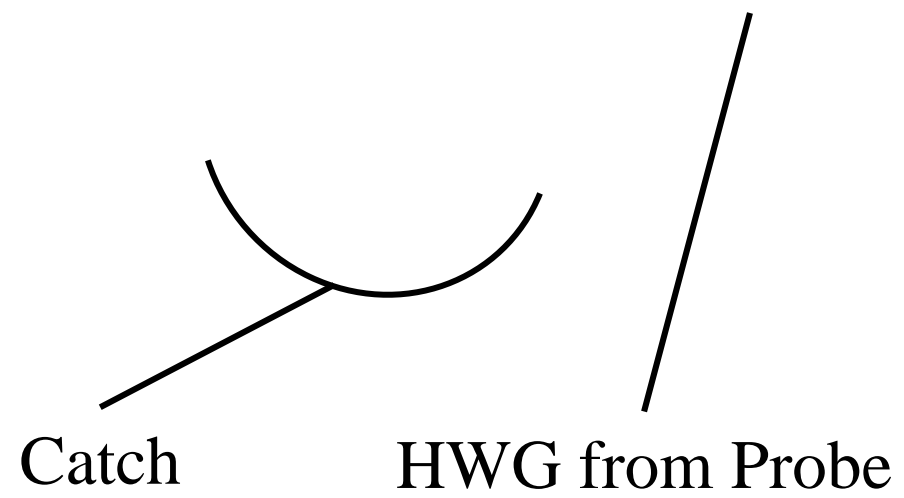


Assembly

HWG to Probe



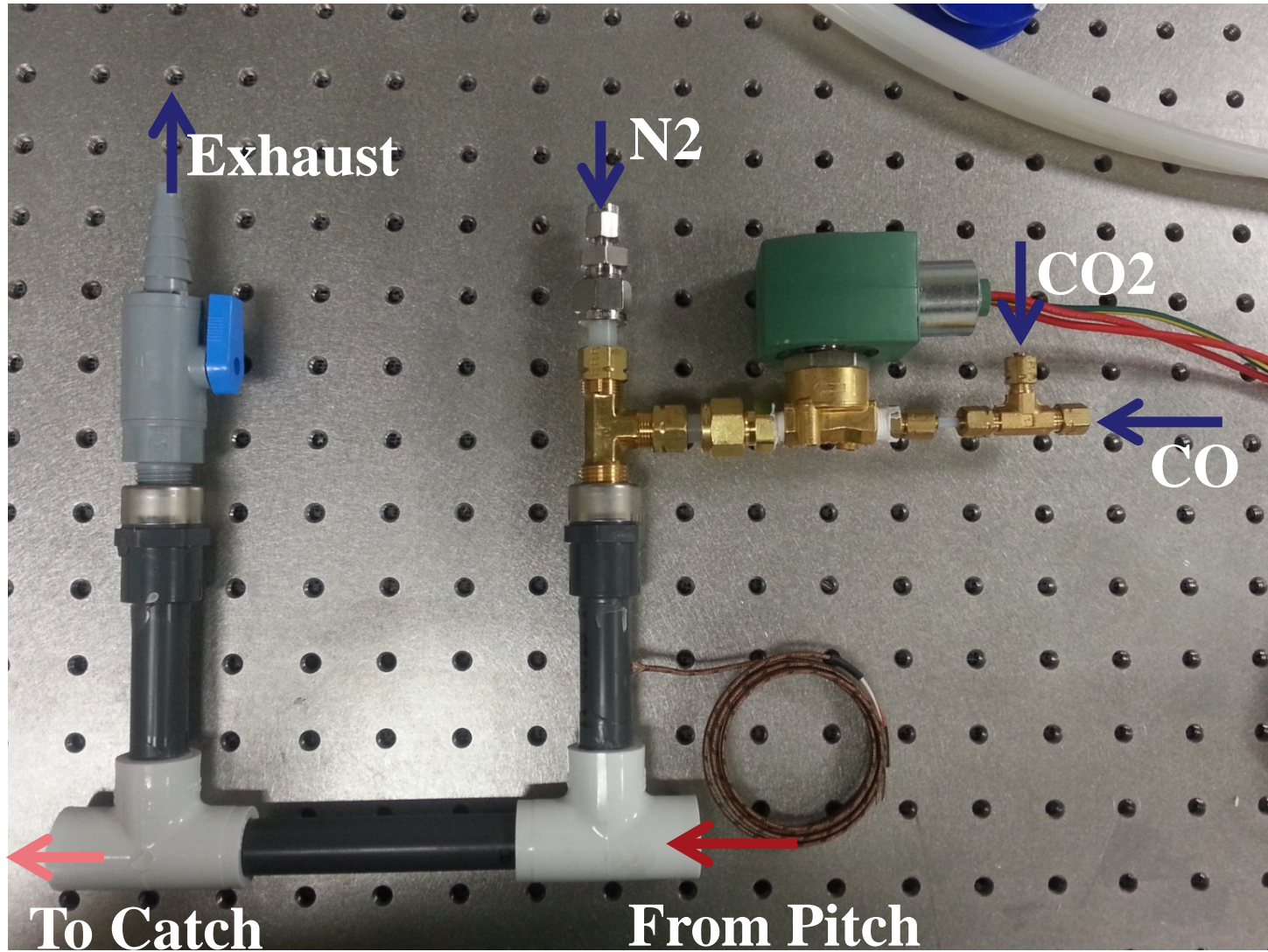
Pitch



Catch

HWG from Probe

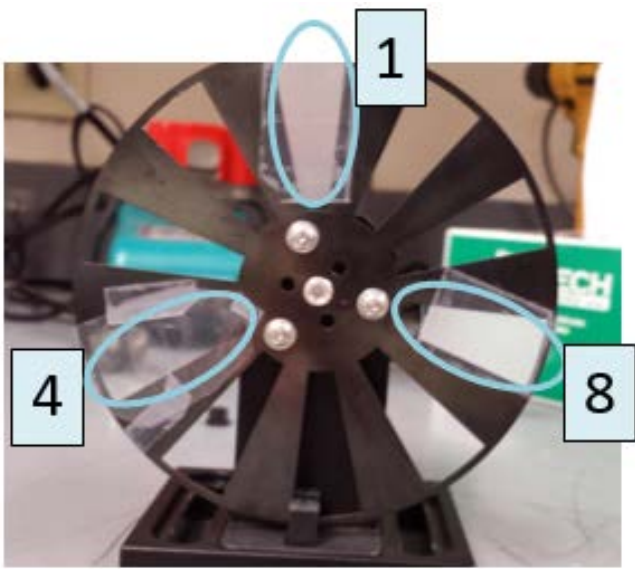
Validating





Detection limit and time resolution characterization

- Early evaluation testing done at ORNL
- Measurements were taken using a flow cell with a path length of 8cm
 - Neat CO₂ measurements
 - Neat CO measurements
 - Simultaneous measurements/evaluation of cross-interference
- Time resolution testing
 - Chopper wheel with plastic to simulate absorption



N₂ 100%

Span Gas

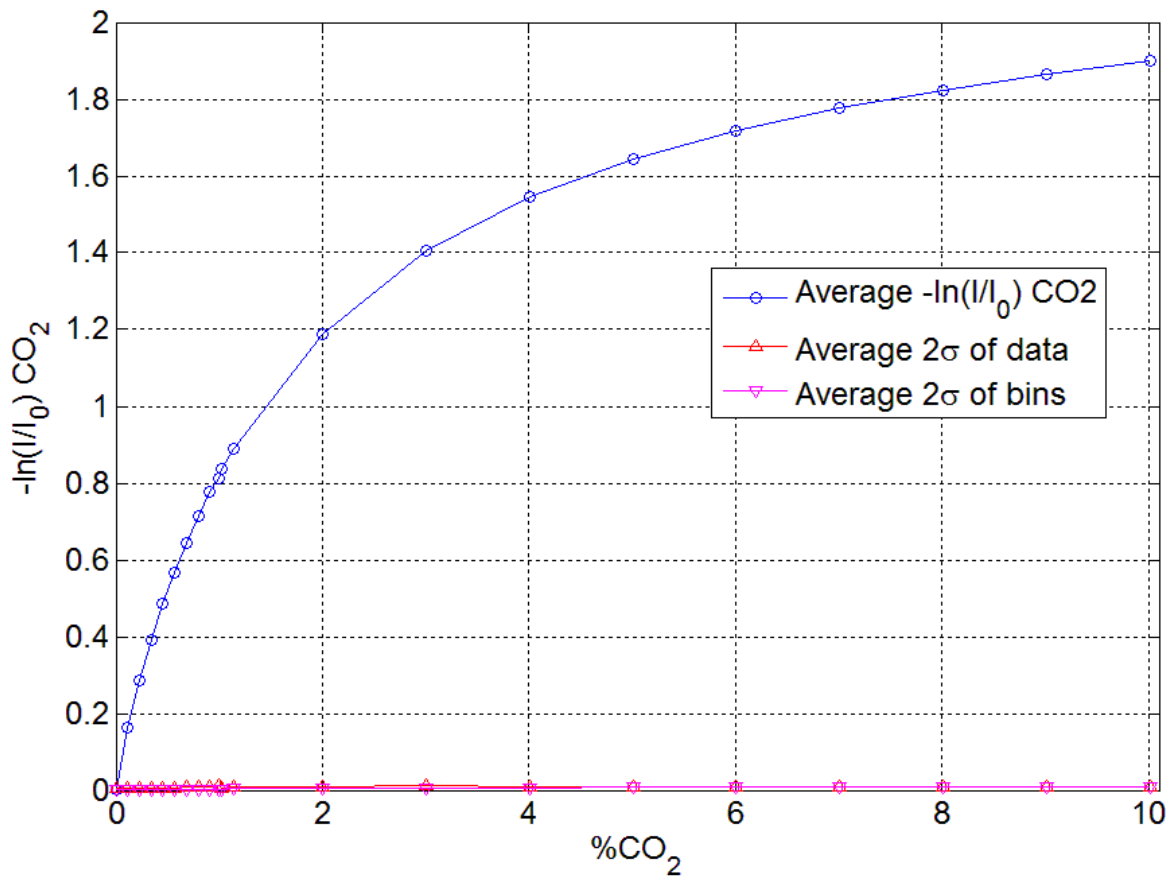


Output to Test Cell



Neat CO₂ Results: 30ppm

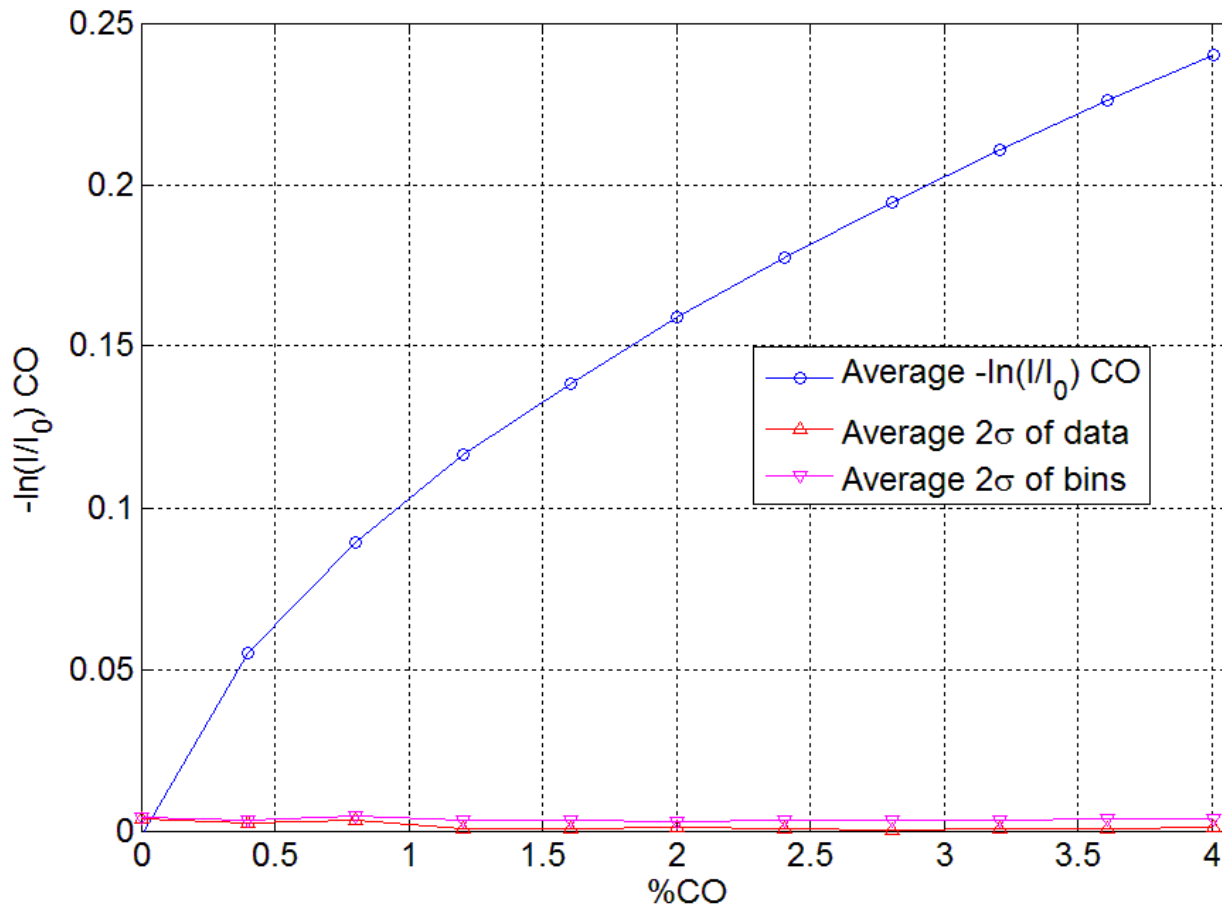
$$-\ln\left(\frac{I}{I_0}\right) = kL\chi$$



Detection Limit: 30ppm (function of path length L)



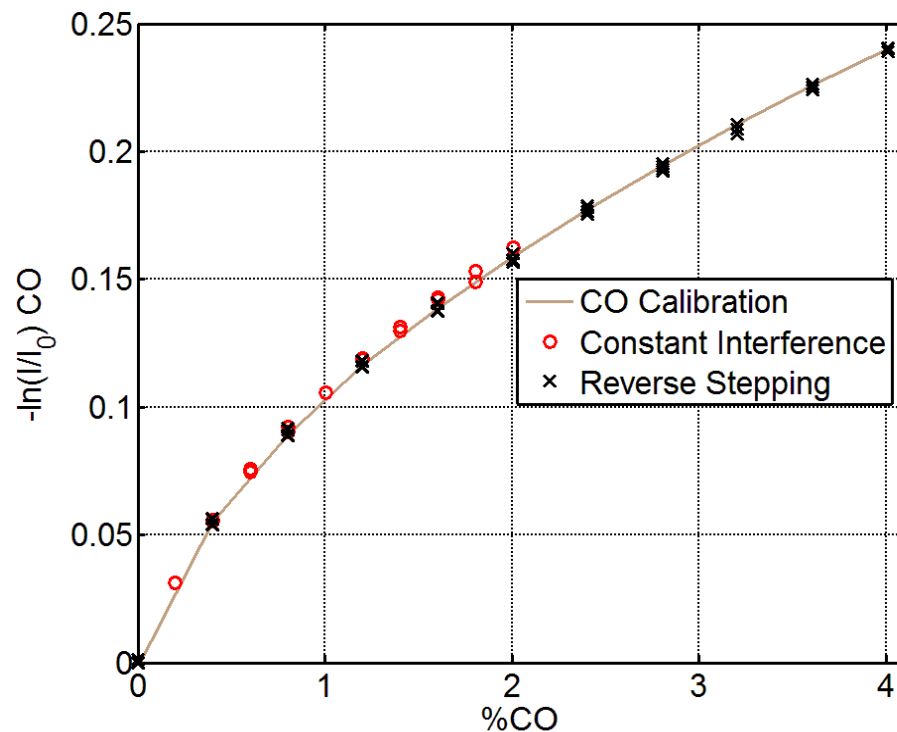
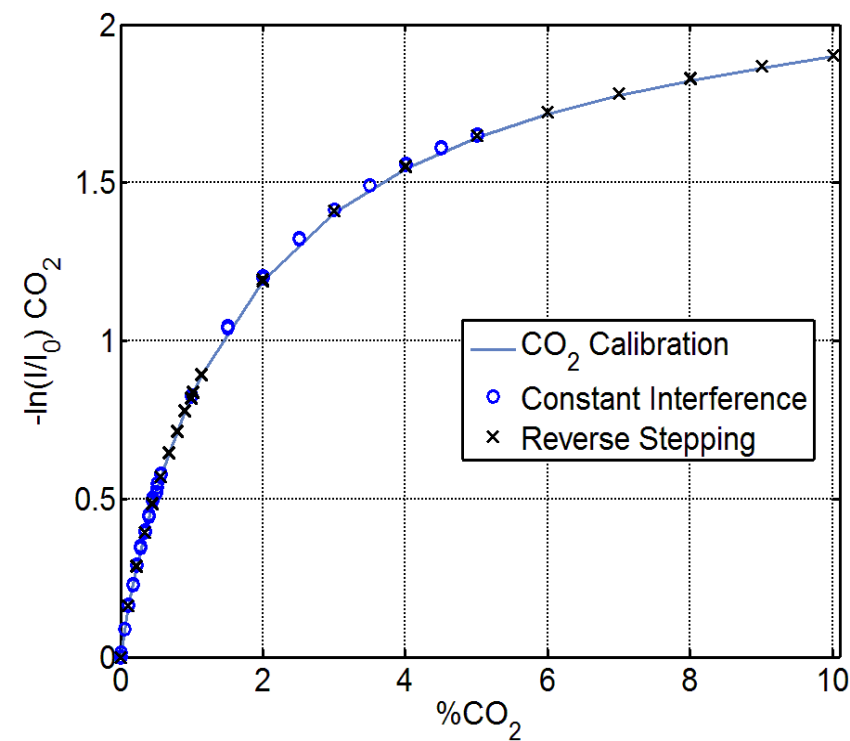
Neat CO Results: 400ppm



Detection Limit: 400ppm (function of path length L)

No cross-interference observed

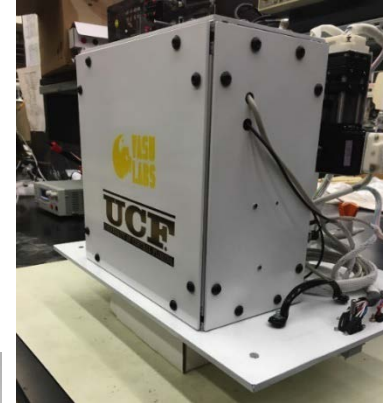
Simultaneous measurements of CO and CO₂





Organization

- ❖ Fundamentals of spectroscopy and absorption technique
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- ❖ **Demo sensor flight test results**





Team Members & Sponsoring Organizations

Principal Investigators

Dr. Subith Vasu
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Collaborators

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Sponsoring Organizations



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University of Central Florida



Erik Ninnemann
University of Central Florida



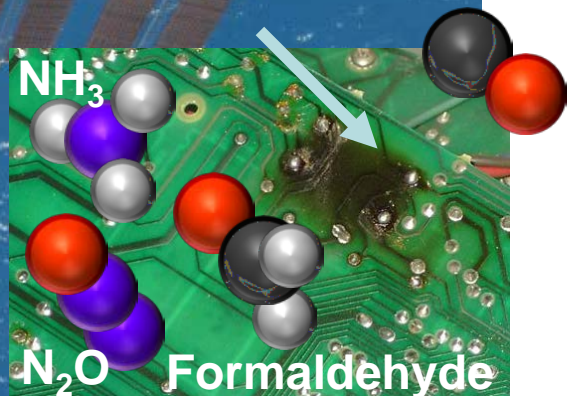
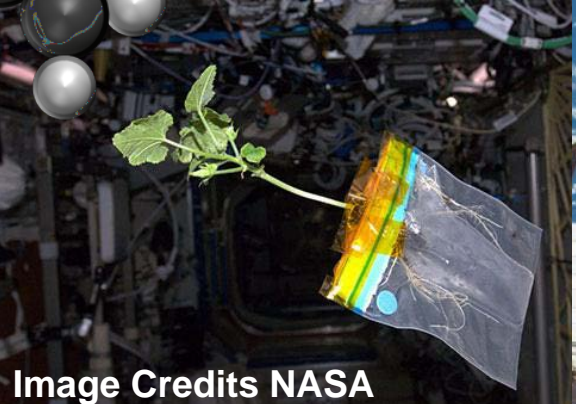
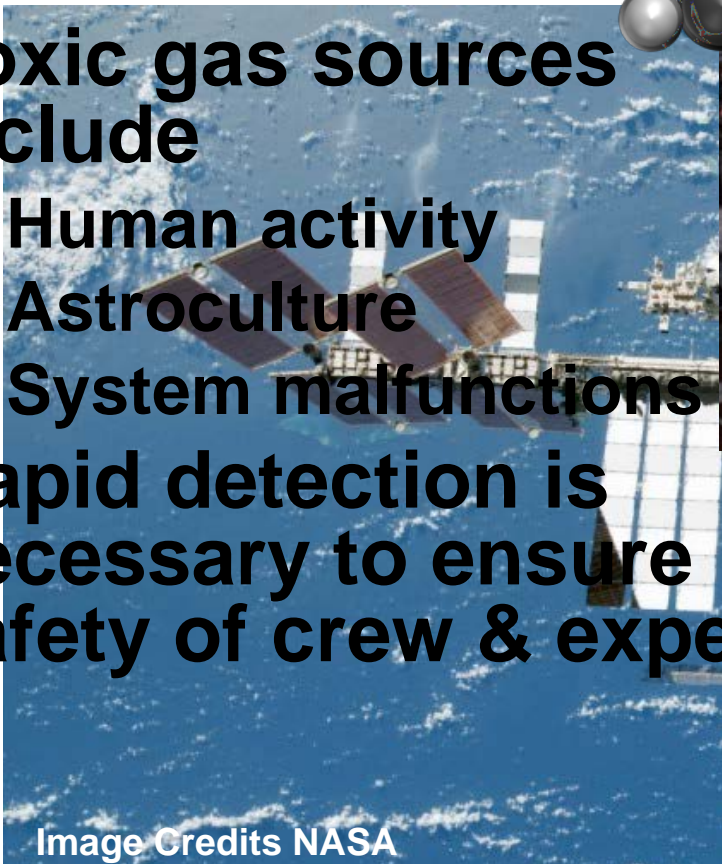
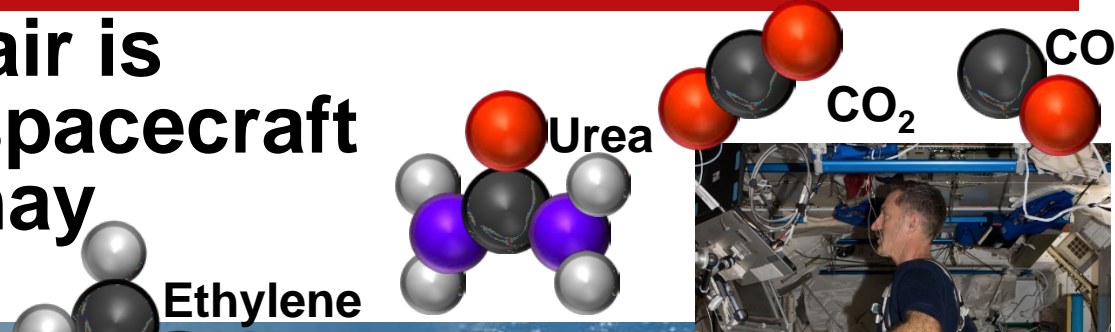
Akshita Parupalli
University of Central Florida



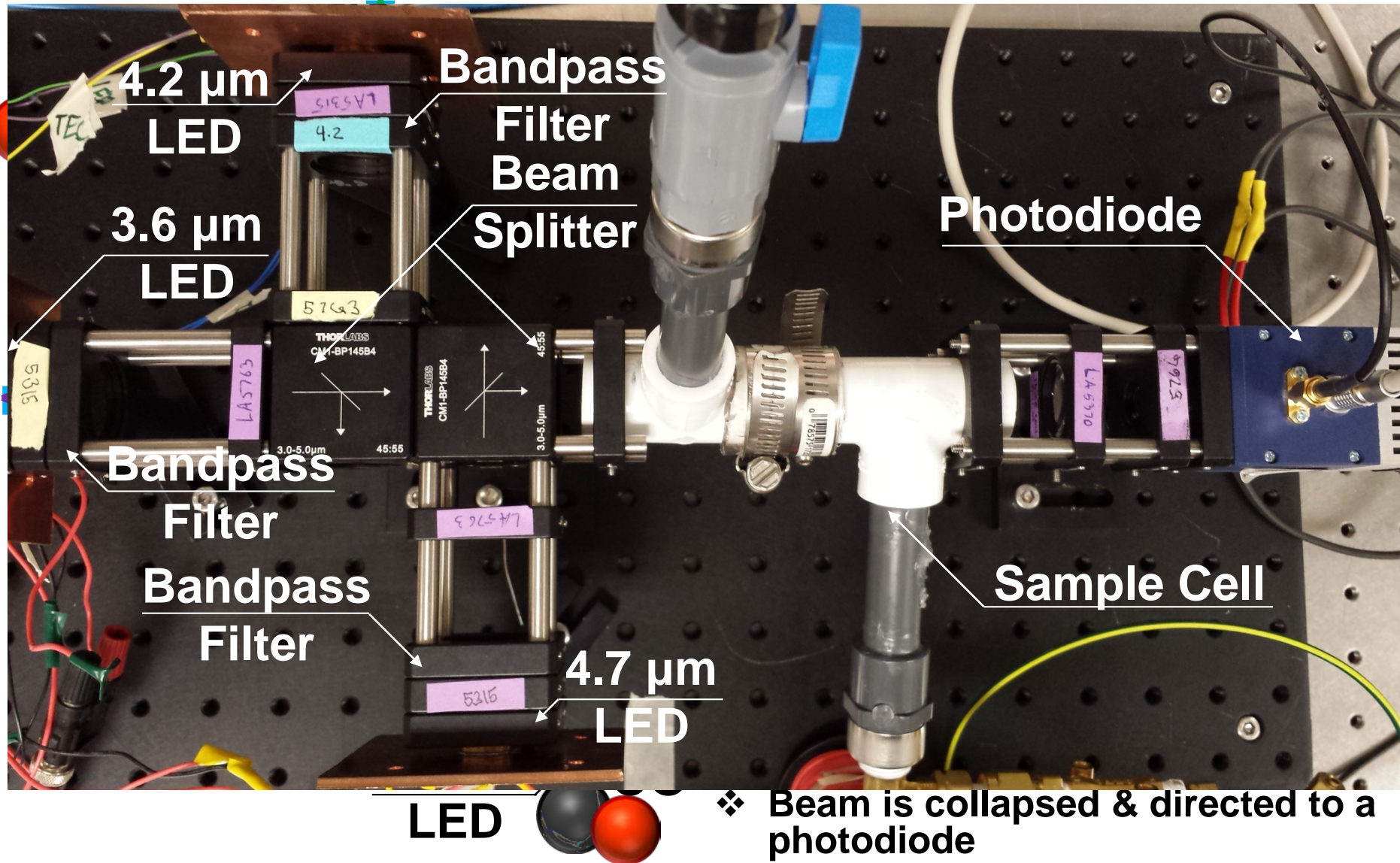


Need for Gas Sensors on Spacecraft

- ❖ Spacecraft cabin air is confined aboard spacecraft and toxic gases may accumulate
- ❖ Toxic gas sources include
 - ❖ Human activity
 - ❖ Astroculture
 - ❖ System malfunctions
- ❖ Rapid detection is necessary to ensure safety of crew & experiments

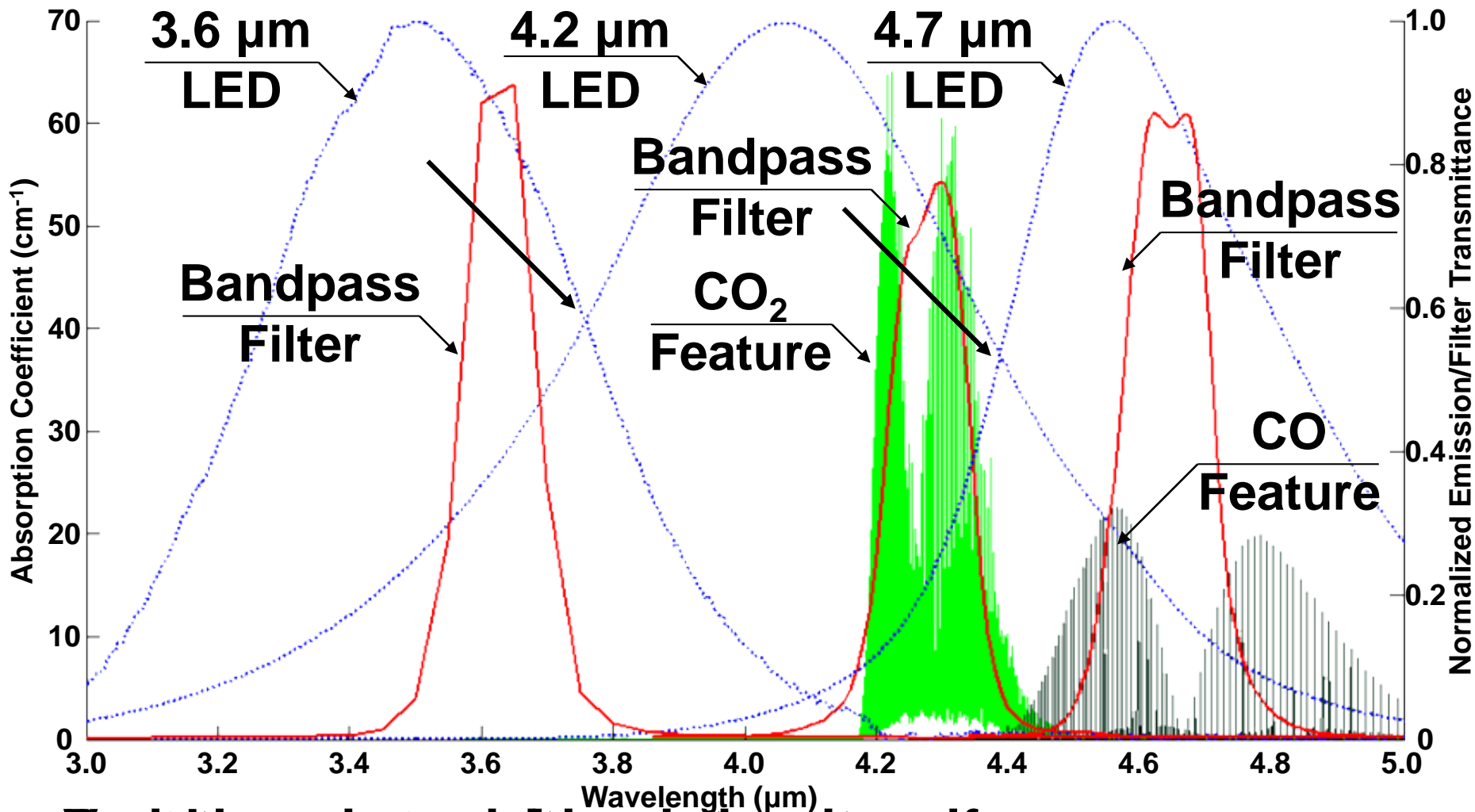


Tabletop Sensor





Spectral Considerations of Tabletop Sensor

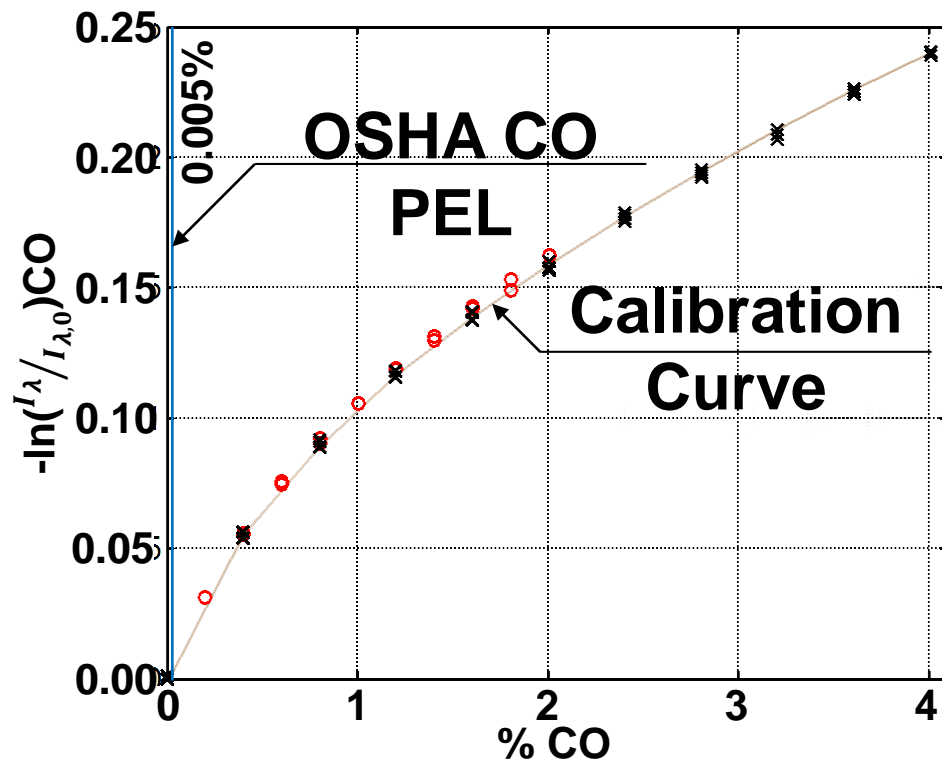
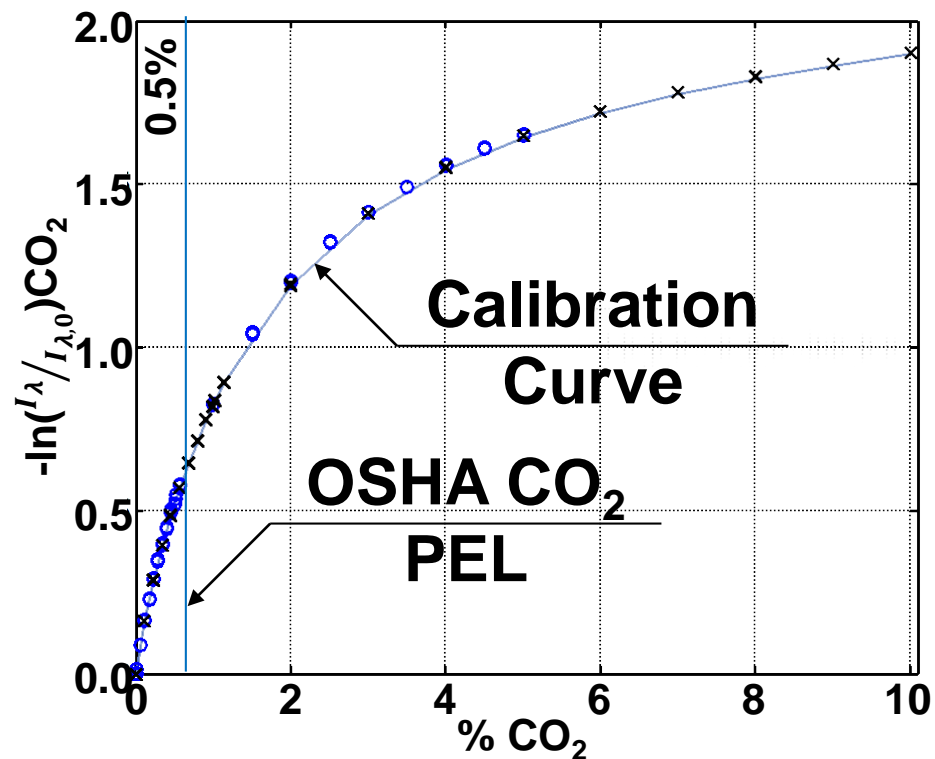


• Spectral power distribution of the LEDs is uniform
 • Light intensity is constant across the LED emission
 • Spectral power is 50 μW, 42 μW, and 50 μW for 3.6, 4.2, and 4.7 μm LEDs respectively
 • Spectral intensity is I_{λ} and $I_{\lambda, \max}$

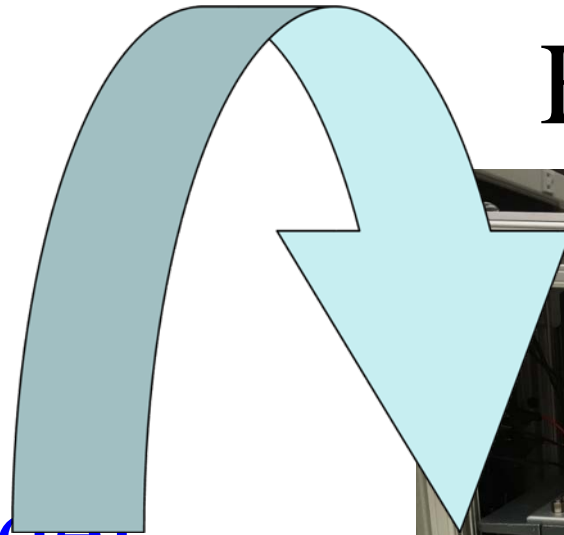


Calibration of Tabletop Sensor

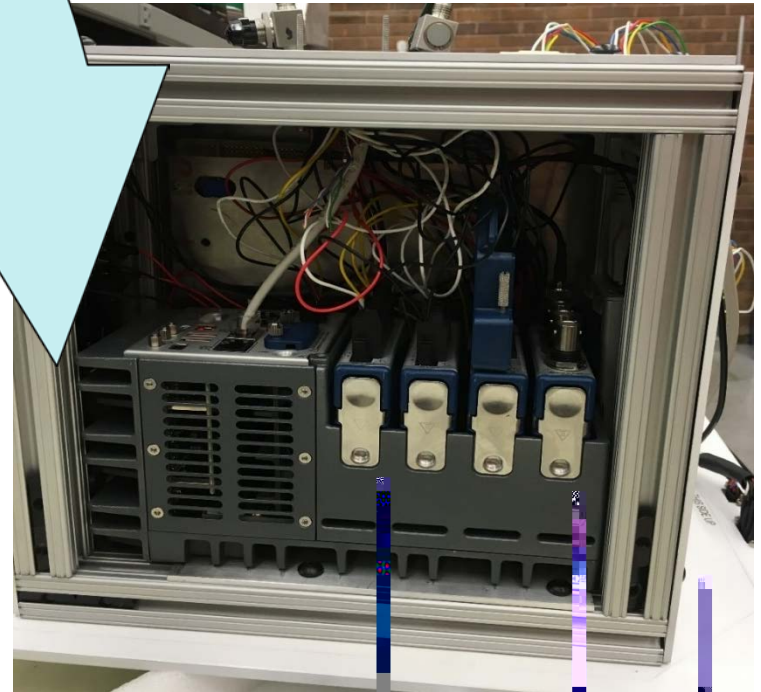
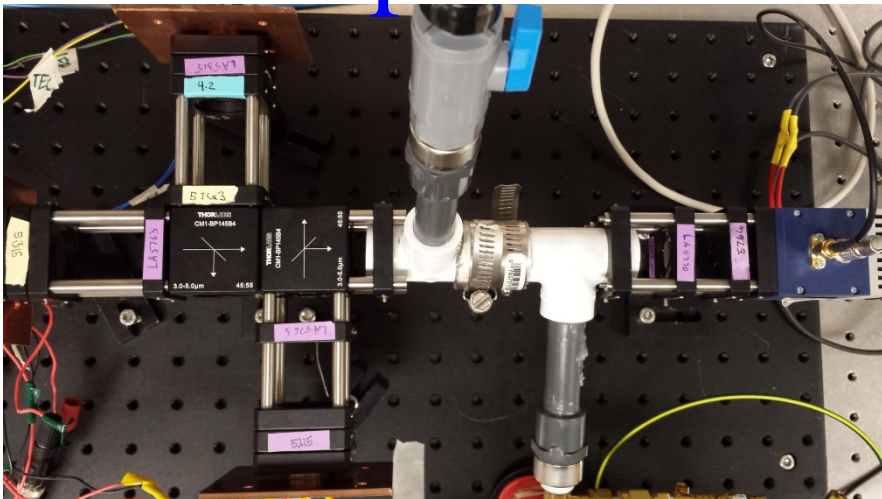
- ❖ Calibration study performed while CO and CO₂ were flowed through a test cell
 - ❖ Test was performed with simultaneous CO/CO₂ test gas
 - ❖ No interference was observed
- ❖ Calibration Curves as shown
 - ❖ OSHA permissible exposure limits (PEL) shown for reference



Flight Test

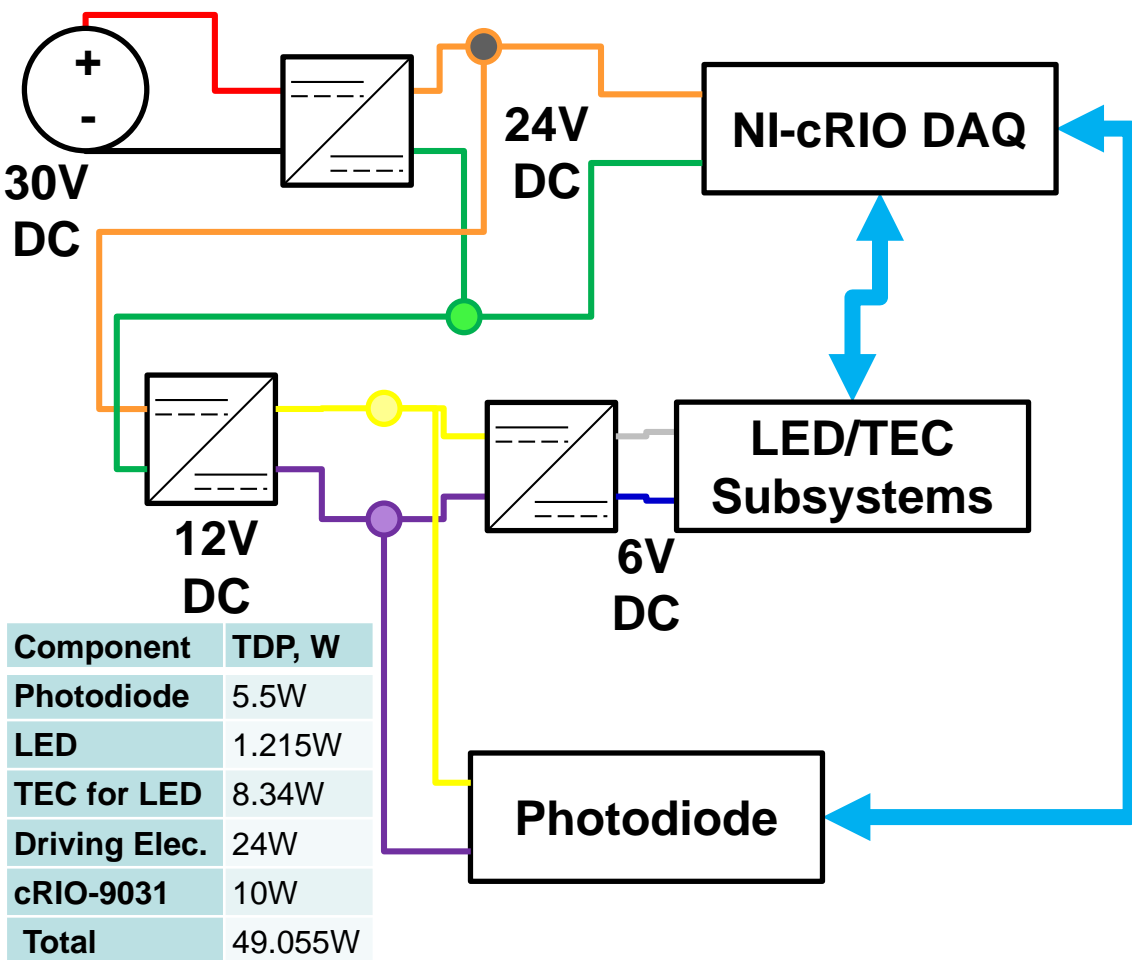


Tabletop Model

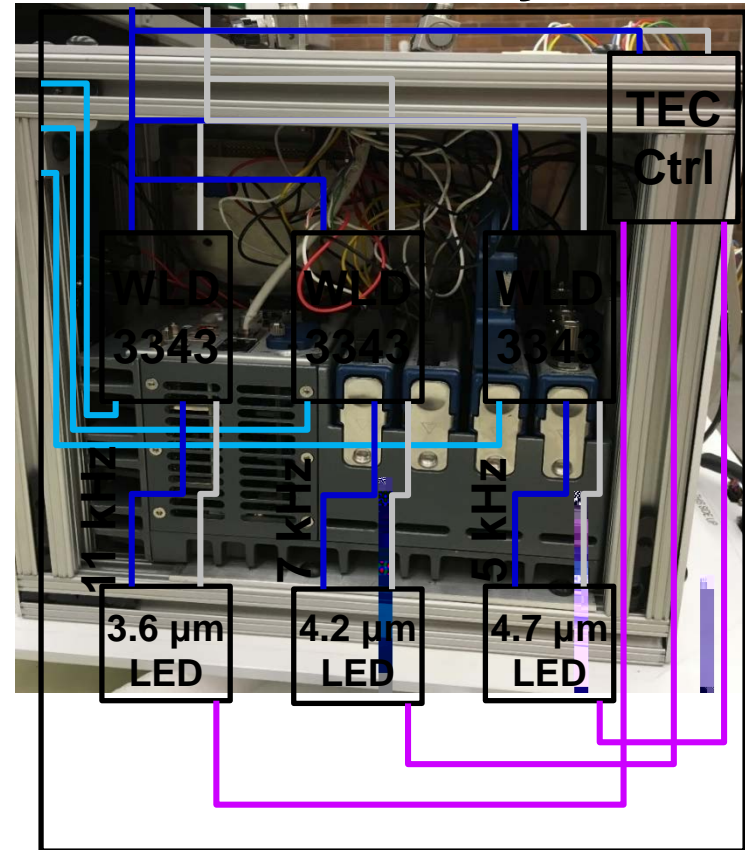




Flight Test Sensor Electrical System



LED/TEC Subsystems



Component	TDP, W
Photodiode	5.5W
LED	1.215W
TEC for LED	8.34W
Driving Elec.	24W
cRIO-9031	10W
Total	49.055W

LED/TEC Subsystem for flight test sensor system
Thermal Design Power by WPT for LED and TEC
Application: Designing a TEC controller



Pre Flight Testing at the UCF Environmental Chamber Testing

- ❖ **Test Conditions**
 - ❖ **Duration 4 hours**
 - ❖ **T_{\min} -20°C**
 - ❖ **T_{\max} 23°C**
 - ❖ **P_{\min} 0.27 kPa**
 - ❖ **P_{\max} 101 kPa**
- ❖ **No observed issues with operation**





High Altitude Balloon Flight Test

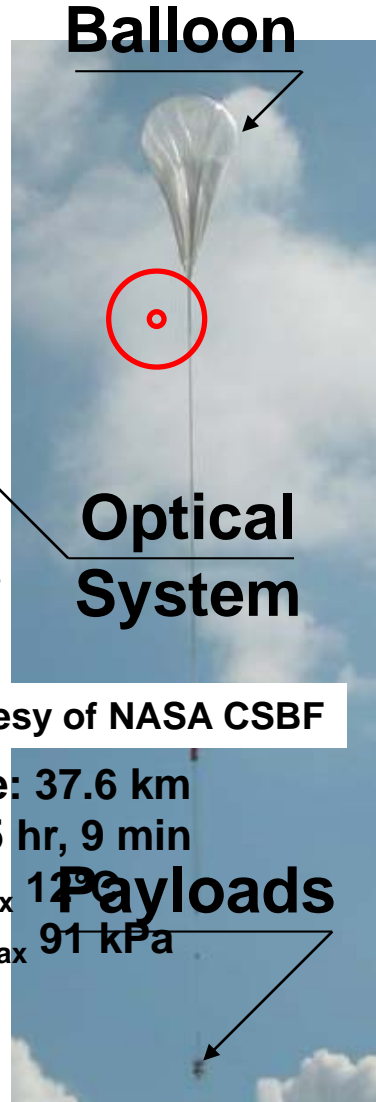
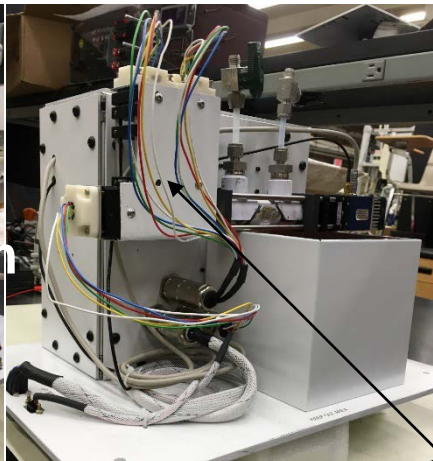
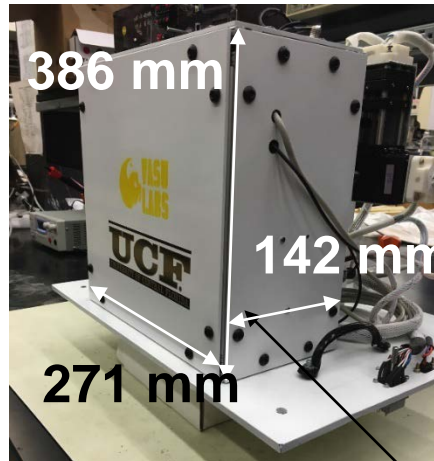
- ❖ NASA Columbia Scientific Balloon Facility Ft. Sumner, NM

- ❖ Sensor packaged and mounted
 - ❖ Aluminum enclosure for sensor electronics
 - ❖ Sensor mounted externally

- ❖ Autonomous operation test
- ❖ Fixed gas composition

- ❖ N₂ 89.51%
- ❖ CO 4.97%
- ❖ CO₂ 5.52%

UCF Payload



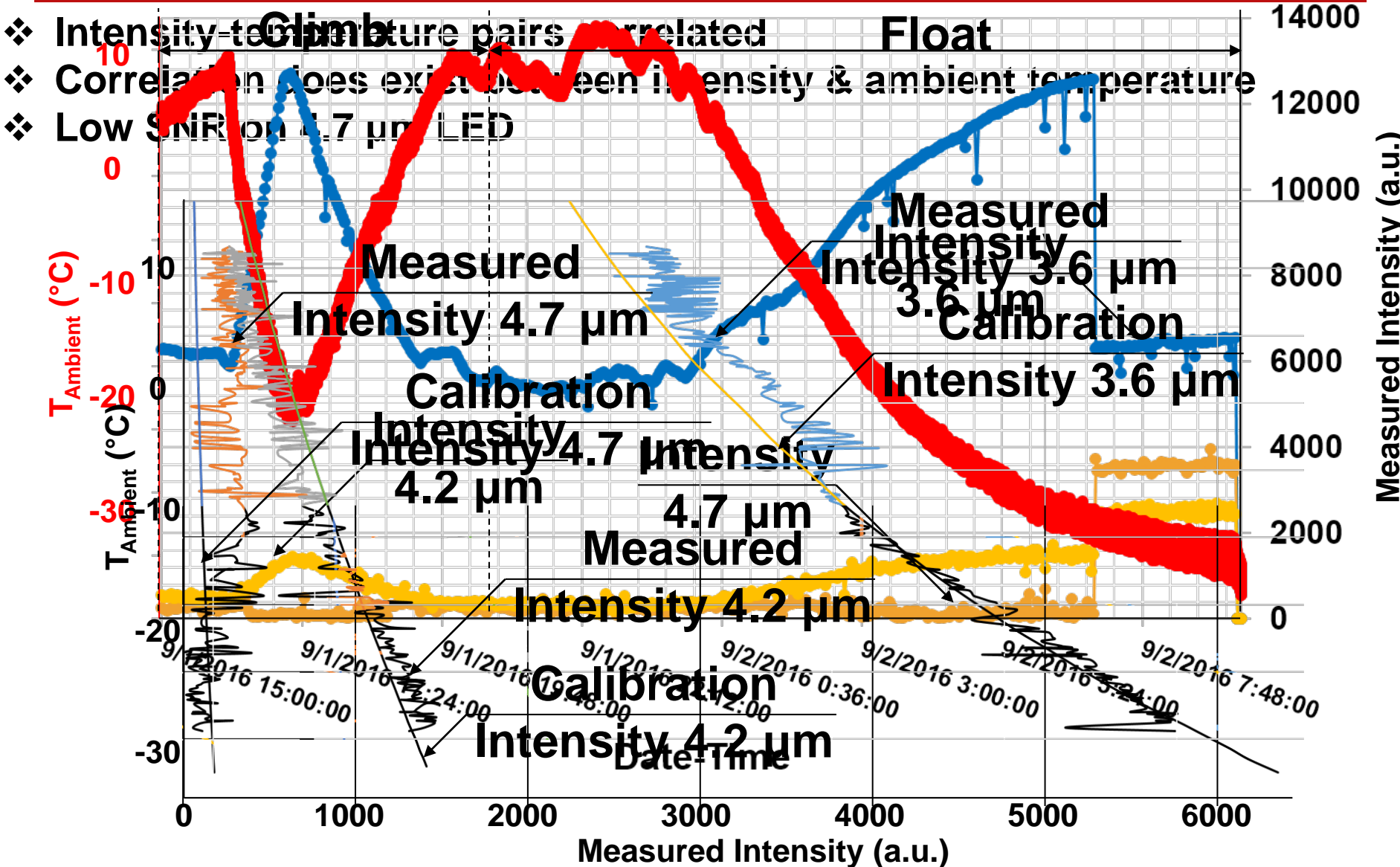
Flight Tracking Courtesy of NASA CSBF

- ❖ Maximum Altitude: 37.6 km
- ❖ Float Duration: 15 hr, 9 min
- ❖ T_{min} -48°C, T_{max} 1°C
- ❖ P_{min} ~0.5 kPa, P_{max} 91 kPa

Payloads

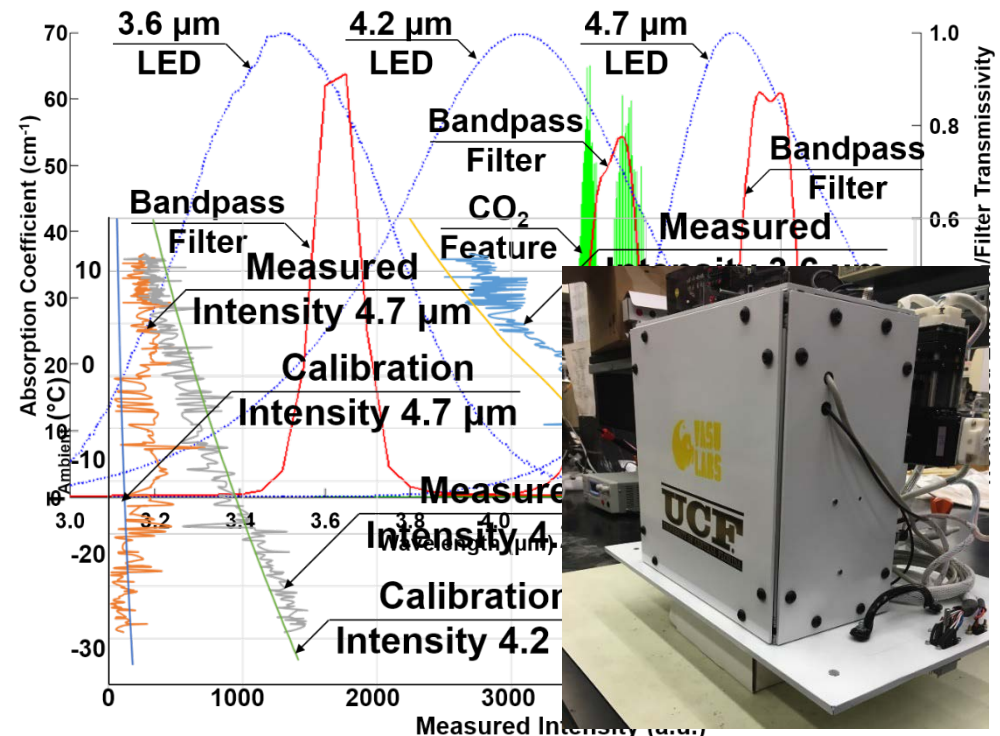
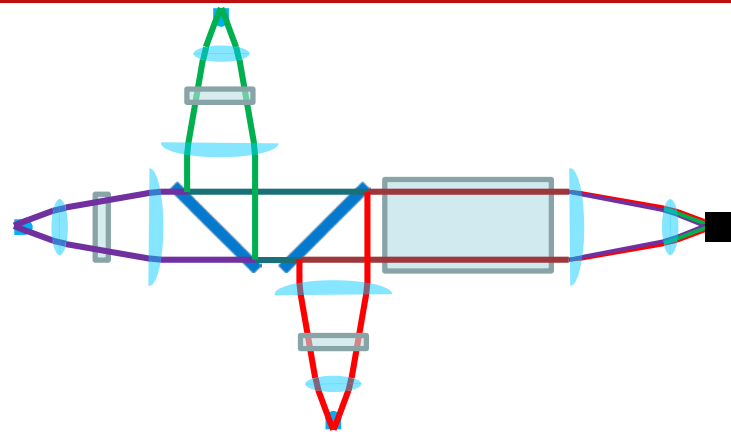


High Altitude Balloon Flight Results



Learned Lessons from Beamsplitter Balloon Flight

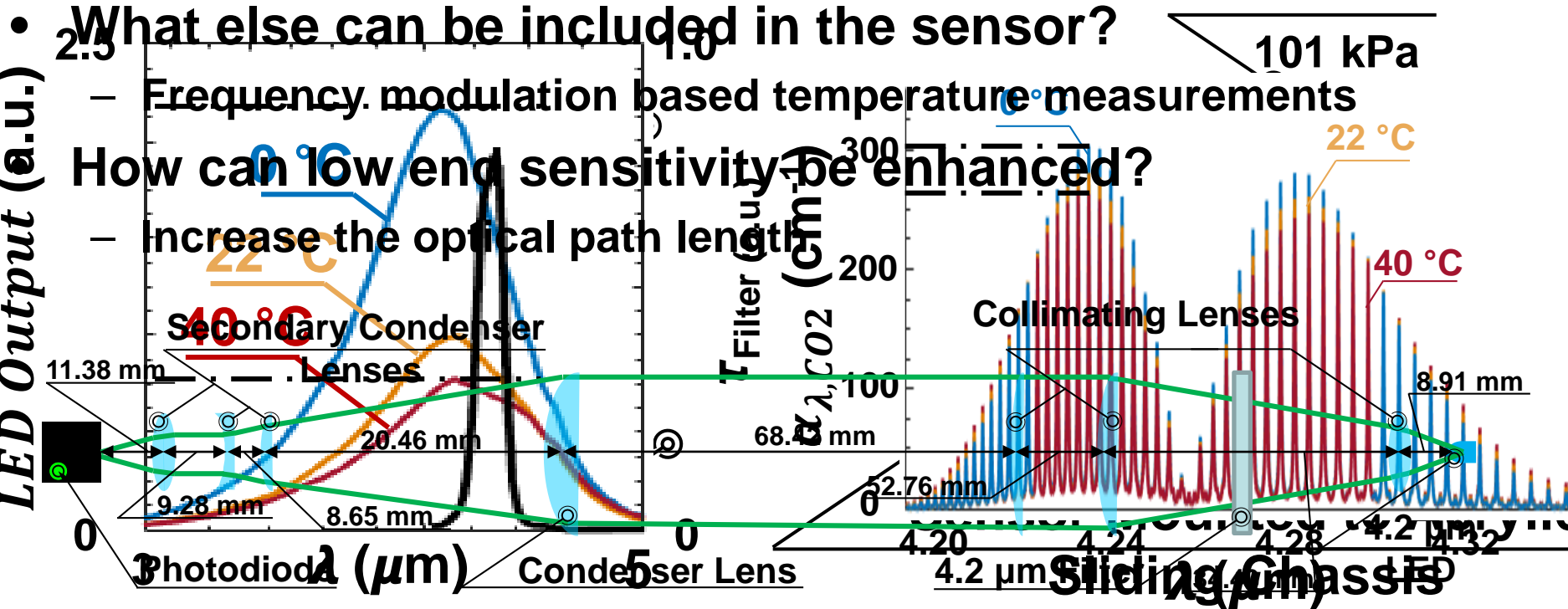
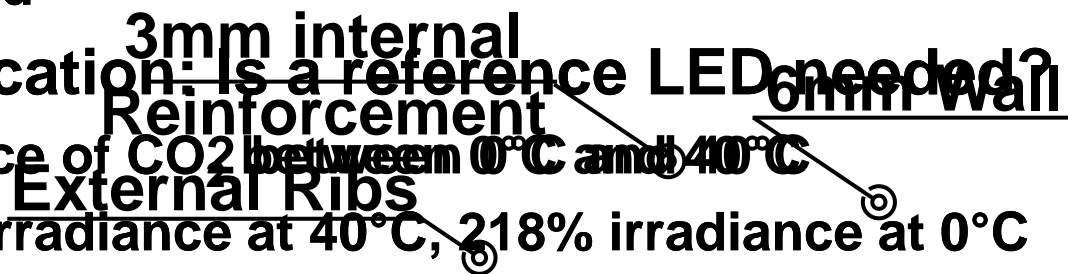
- ❖ **Low SNR**
 - ❖ 45% Intensity loss at each beamsplitter
 - ❖ Test cell is too small
- ❖ **Band pass filters limit versatility**
 - ❖ Poor overlap of CO feature with 4.7 μm filter
 - ❖ Gases outside of bands are nondetectable
- ❖ **High signal variance from calibration**
 - ❖ LED temperature regulation was sub par
 - ❖ Nested ground loops caused increased noise
- ❖ **Sensor is too big**
 - ❖ High TDP
 - ❖ Large volume
 - ❖ Heavy



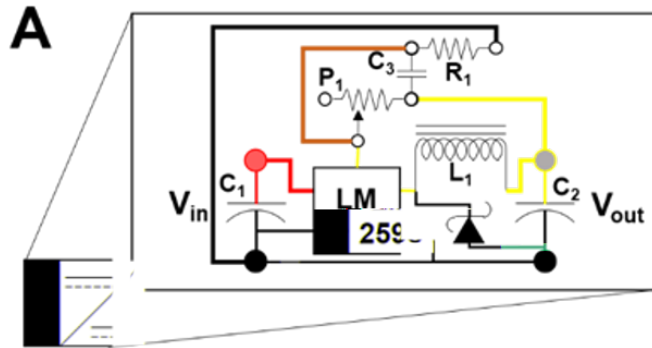


Rethinking & Improving the Sensor Design

- **Weight Reduction: Laser cut acrylic enclosure**
 - Test cell no longer needed
- **Optical System Simplification: Is a reference LED needed?**
 - 15% change in absorbance of CO₂ between 0°C and 40°C
 - If 22°C is nominal; -25% irradiance at 40°C, +18% irradiance at 0°C



Condensed Electrical Systems



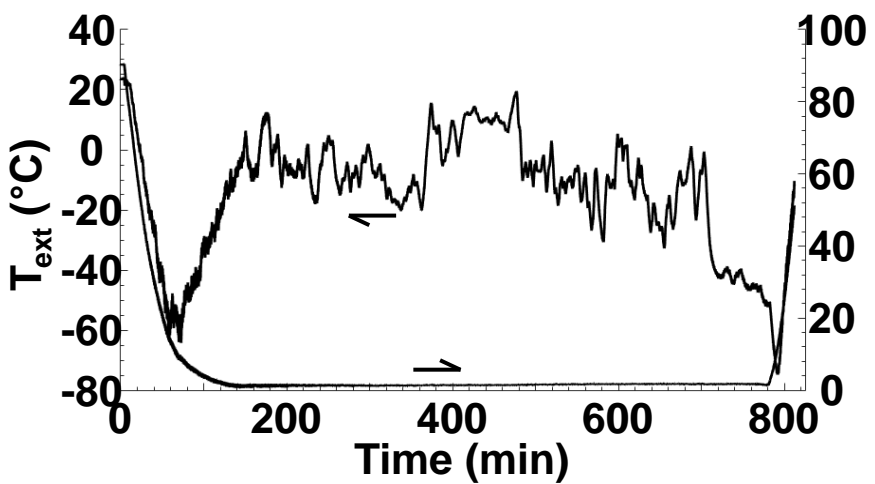
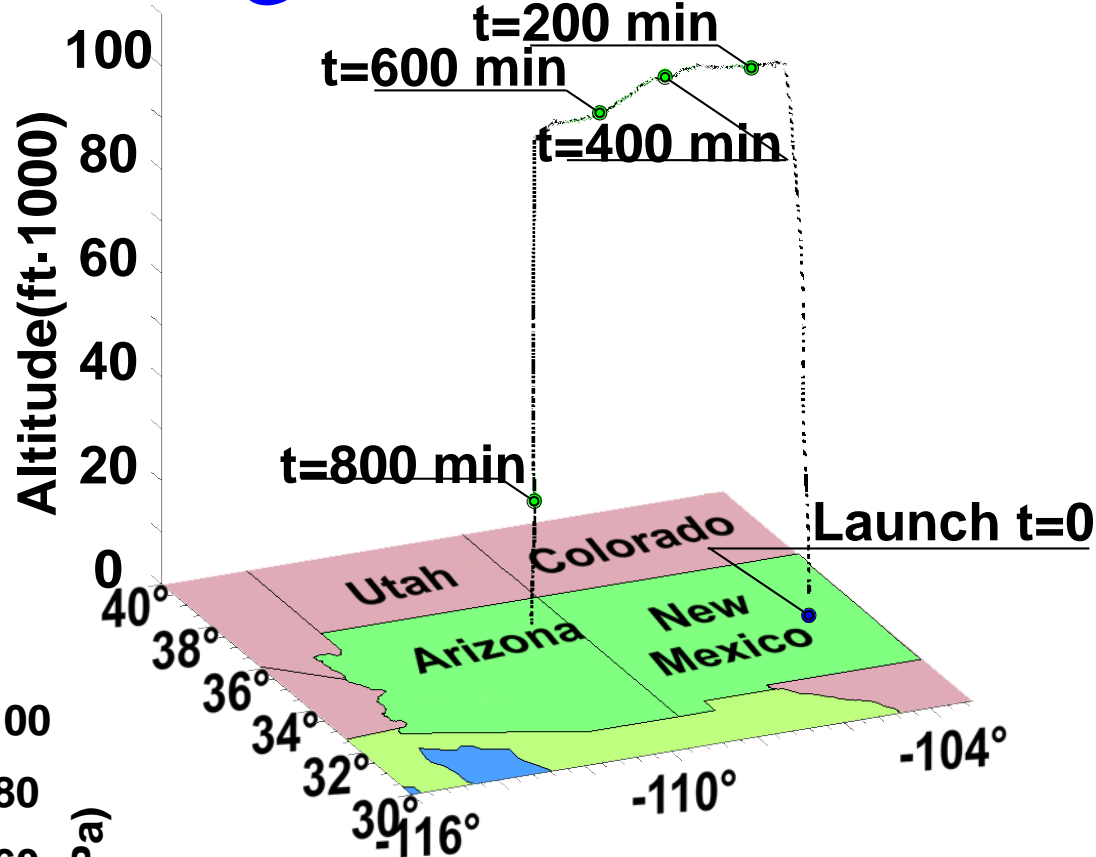
Legend			
C_1 100 μ F	C_5 3.3 mF	L_1 33 μ H	P_1 30 k Ω
C_2 220 μ F	C_6 10 nF	R_1 1 k Ω	P_2 50 k Ω
C_3 33 nF	C_7 1 nF	R_2 200 k Ω	
C_4 330 μ F	C_8 100 nF	R_3 100 Ω	

Maximum power draw reduced to 17W, 10.6 of which is cRio



High Altitude Balloon Flight Test

- Launch at Ft. Sumner 9/4/17
 - 138 min ascent to Alt_{Avg} 105,000'
 - Float for ~10.5 hrs
 - Alt_{Max} 110,000'
 - Total flight time 820 min
 - P_{min} 1.6 kPa, P_{max} 90 kPa
 - T_{min} -64°C, T_{max} 23°C

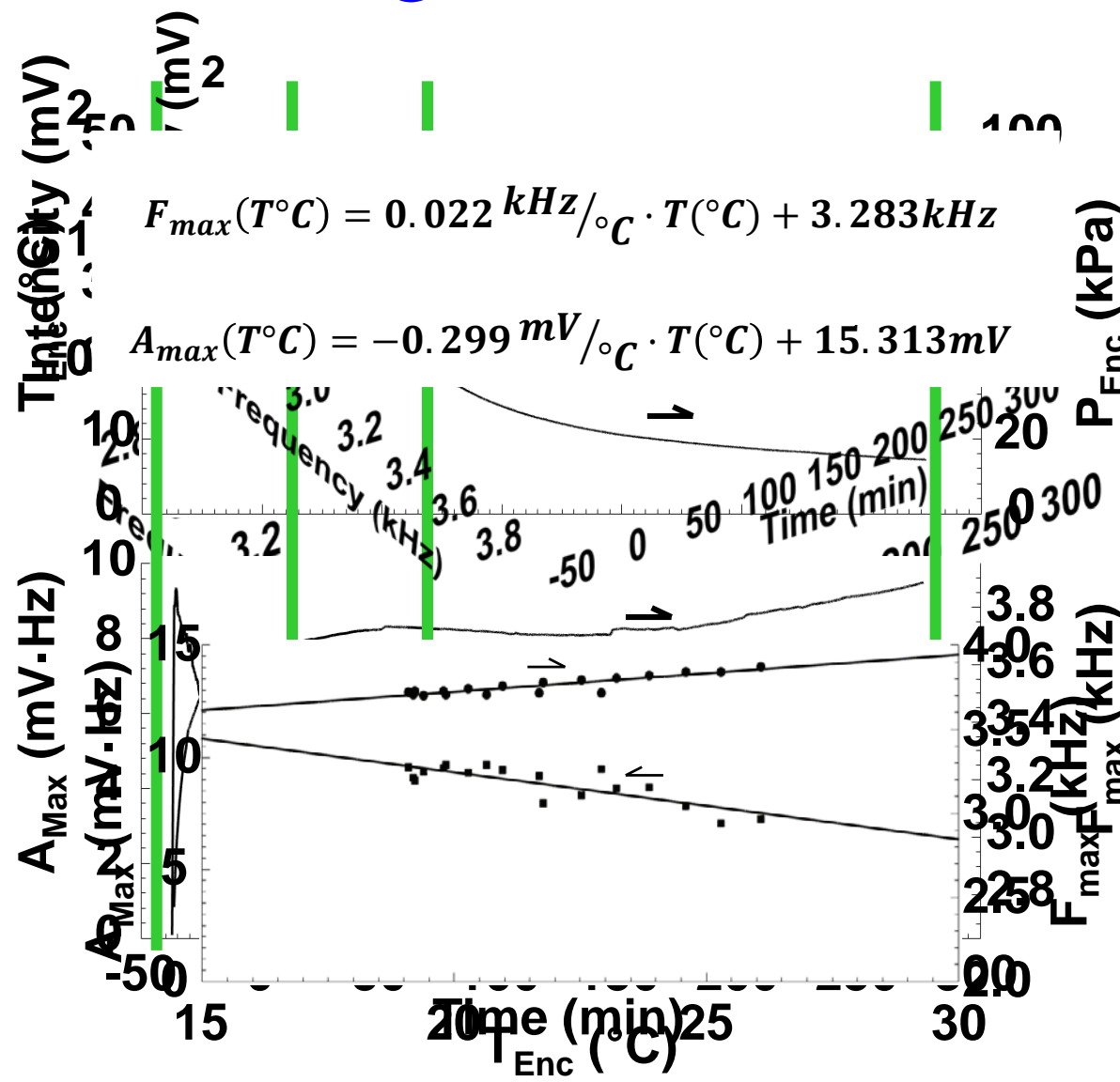


- Air at 90 kPa packaged into the enclosure with sensor
 - Fixed CO_2 concentration
- Flight testing examines signal drift and noise floor under real conditions



Condensed Flight Data

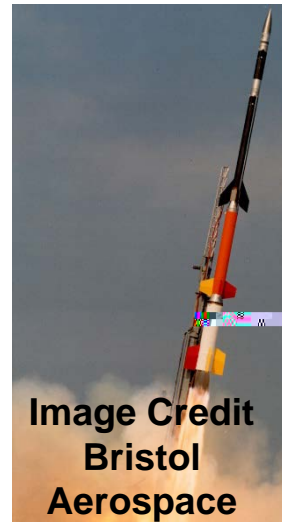
- **Power Spectra**
 - 2¹⁶ data points sampled at 100 kHz
 - FFT applied to each sample set
- **Locations of maxima correspond to CO₂ measurements**
- **F_{max} and A_{max} are both temperature dependent**
 - F_{max} transient at startup
- **After, ↑ F_{max} and T_{Enc} ↑**
 - ↓ A_{max} envelope and T_{Enc} ↑
- **Linear relations are found for F_{max} and A_{max} under constant CO₂**





Conclusions & Future Work

- ❖ LED based gas sensor is to be used for the detection of toxic compounds in spacecraft air
- ❖ Both amplitude and frequency modulation are simultaneously used in the sensor
- ❖ Next iteration will use multiple LEDs and variable path optical system to enhance species selectivity and LDL
 - ❖ MHz modulation instead of kHz
- ❖ Sounding rocket test





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Vasu Lab

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Publications for further reading

- Anthony Terracciano; Kyle Thurmond; Michael Villar; Justin Urso; Erik Ninnemann; Akshita Parupalli; Zachary Loparo; Nick Demidovich; Jayanta S. Kapat; William P. Partridge, Jr.; **S. Vasu**, “Hazardous gas detection sensor using broadband LED based absorption spectroscopy for space applications”, *New Space*, 2018, 6 (1), 28-36. **Cover Page Article**.
- Kyle Thurmond; Zachary Loparo; W.P. Partridge Jr.; **Subith S. Vasu**; “A Light-Emitting-Diode (LED) Based Absorption Sensor for Simultaneous Detection of Carbon Monoxide and Carbon Dioxide”, *Applied Spectroscopy*, 2016, 70 (6), 962-971.



Questions?
