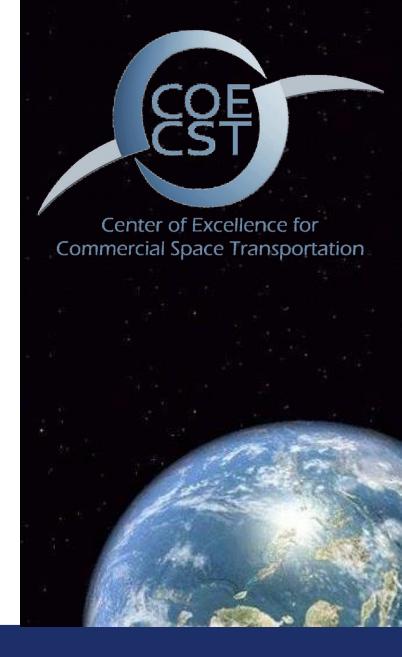
#### **COE CST Tenth Annual Technical Meeting**

Tasks 323
Structural Health Monitoring (SHM)
Framework for Commercial Space
Transportation:

Portable electro-mechanical impedance SHM hardware unit development and design of spacecraft data acquisition system as a precursor of future flight recorder

Andrei Zagrai (NMT) & Dale Amon (Immortal Data)



#### **Team Members**

- People
  - Andrei Zagrai and Dale Amon





- Students ... in progress of getting visa.., Covid19 impact
- Organizations
- New Mexico Tech and Immortal Data, LLC
  - Both providing matching funds

## Why?

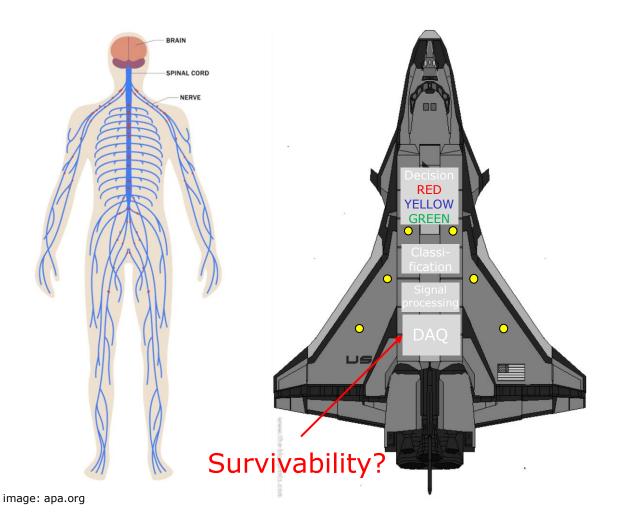
- Improve safety
- Reduce operation cost
- Structural Health Monitoring (SHM) is seen as one of key technologies that could allow to achieve this goal.

A loose pin prevented one of the parachutes tucked inside Boeing's CST-100 Starliner capsule from deploying during November 4, 2019 crewless abort test in New Mexico.



https://aerospaceamerica.aiaa.org/starliner-descendsone-chute-short-but-nasa-declares-success/ Image credit: NASA

## **Spaceship that Feels!**



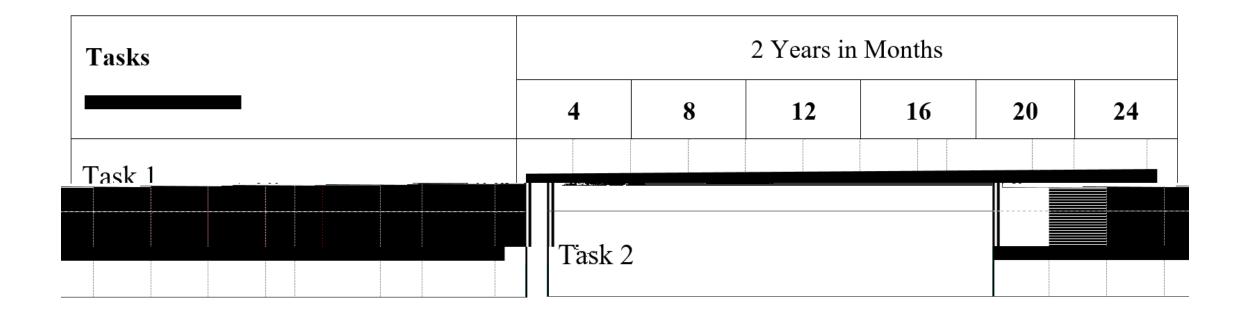
- Passive and active sensors
- Local / global data acquisition
- Signal processing and feature extraction
- Feature classification and automatic decision support (Bayesian, Neural Net)
- Traffic light output

RED, YELLOW, GREEN

### **Task Description**

- Focus is:
- Development of a portable multi-channel electromechanical impedance SHM hardware (NMT lead) and
- Design of spacecraft data acquisition system as a part of SHM framework for CST vehicles, payloads and components. (Immortal Data lead)
- We intend to demonstrate results of development in tests flight(s) on commercial space vehicle(s).

### **Schedule**

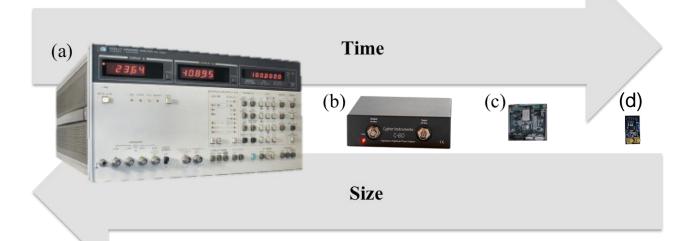


### Goals

- Develop a spacecraft data acquisition system which enabled data survivability during catastrophic event
- Develop a small/lightweight electromechanical impedance structural health monitoring (SHM) unit
- Propose integration of SHM and data acquisition system to improve flight safety and reduce flight costs.

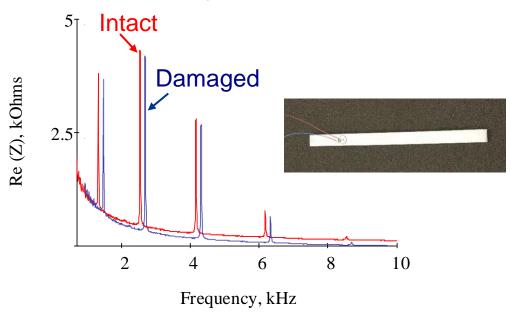
### Miniaturization of Impedance Analyzers

- Historically, impedance analyzers were heavy and bulky
  - E.g. HP 4192A impedance analyzer
  - 43 x 24 x 62CM, 19kg



- (a) Hewlett Packard 4192A, (b) Cypher Systems C-60
- (c) LANL WID3, (d) NMT Canary

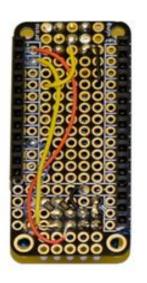
- Technological progression
  - PC and DAQ based systems
  - Microcontroller and FPGA based systems.
    - LANL WID system in 2008
    - NMT's Canary 2018



### **Impedance Measurement Unit - Canary**











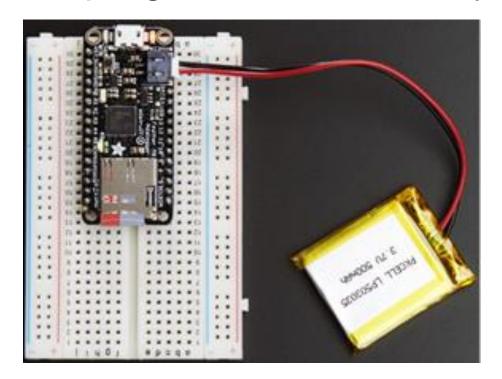


- Full control of AD5933
  - Adjust voltage, gain, and settling time
  - Set frequency in Hz or by frequency code
  - Frequency codes provide 1/32 Hz resolution

- 1kHz to 100kHz specified frequency range
- Save data to micro SD card
- Some post processing of data capability

### **Battery – powered Unit**

 AD5933 + Adalogger = A portable, battery powered, programmable SHM system

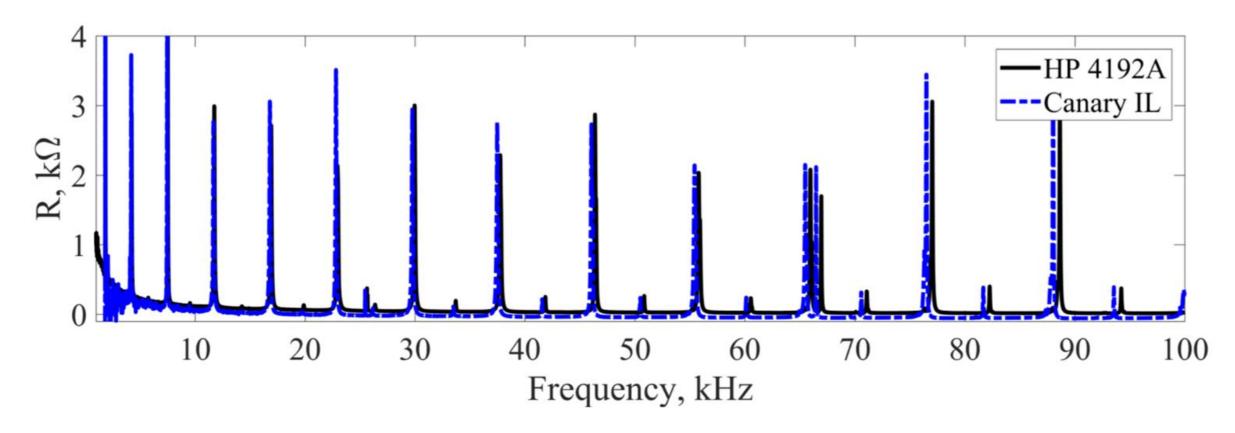




Measures 4cm x 3cm x 0.5cm and weighs 10g

### Electromechanical Impedance SHM

Real part of impedance of metallic structure (6061-T6 aluminum circular plate ) over broad frequency band



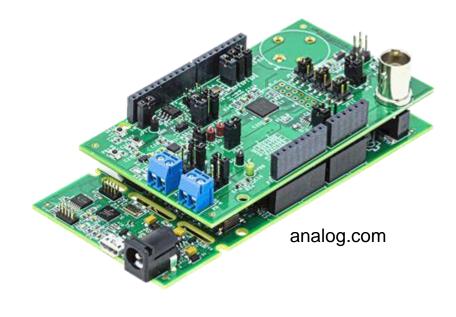
### Single-channel unit progress



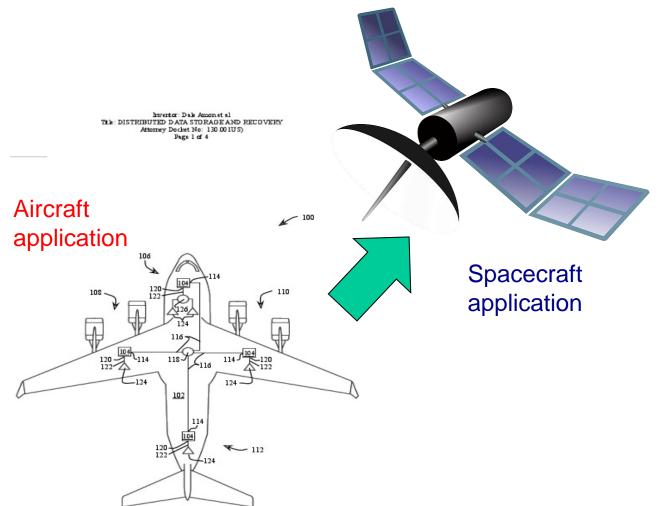
- Dyllian Powell has provided parts inventory, operation manual and prepared portable computer system to work with Canary.
- We intend to build 3 single-channel Canary units for integration with Immortal Data distributed black box demonstration suborbital flight.
- Ordering of parts for 3 units is in progress

### **AD5941 New Chip Consideration**

- AD5941 is a new generation of chip allowing for impedance measurements.
- Frequency band is improved from 100 kHz to 200 kHz and even possibly 400 kHz.
- Internal clock improvements from 16 MHz to 32 MHz which will allow for improvements of measurement accuracy.
- Improvements in signal-to-noise ratio from 60 dB to 80 dB



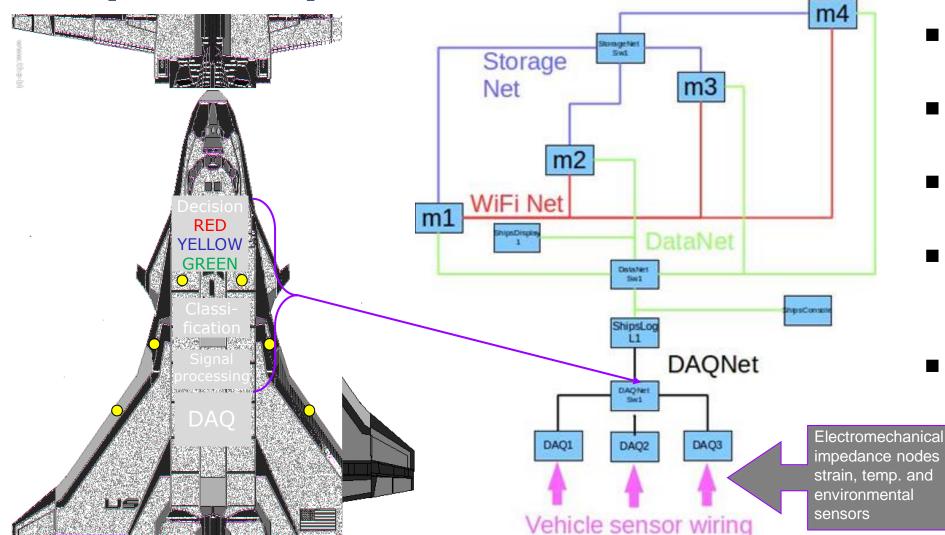
#### Distributed Black Box™



PATENT #: 9286738 USPTO DN/20150339867

- An Immortal Data patent.
- Numbers works for insects, so why not data? You just can't kill them all.
- Nodes can trade GPS data during the first few seconds after a breakup.
- Every unit knows the trajectory of each of the others.
- Recover any single unit and you have nearly all the data to the last milliseconds.
- Useful in a variety of environments including aircraft as well as spaceships.

Spaceship SHM and Data Survivability



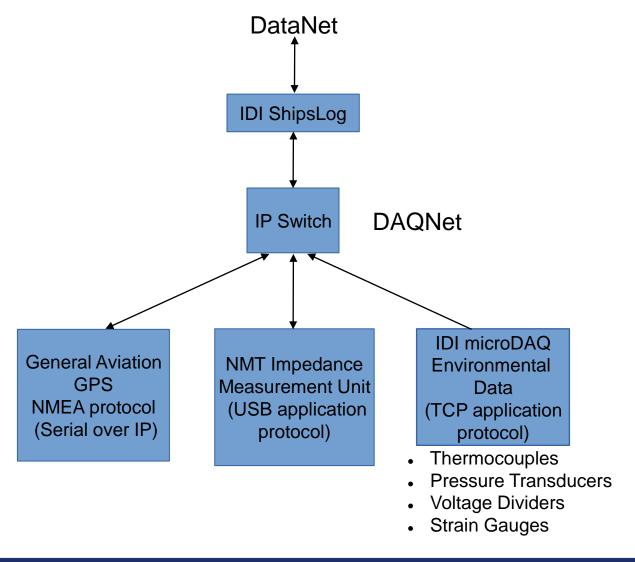
- Redundant path and storage is a key
- ShipsLog™ node polls DAQs on DAQNet
- Data stream is distributed over the DataNet
- Displays, downlink converters, and etc. reside on the DataNet.
- ShipsStore™ can use its private StorageNet and
   WifiNet to share the data.

### Software Design Philosophy

- Hand tooled Real Time loop to collect data, running at a high real time priority in RT Linux.
- Vendor agnostic structure. Only needs a plug in written to interface a new real time data source. We have tested this with a UEI DAQ and an NMEA GPS as data collection peripherals.
- True Real Time is not just optimized use of CPU cycles; it requires traffic on IP networks must be high speed, partitioned, limited to specific uses and traffic tightly controlled to keep sample dropouts at a very low level.
- Standardized low rate / high rate UDP data streaming protocol with only a simple TCP subscription protocol for initial set up.
- Interface to other vendors and other protocol standards is isolated via a translator compute module.

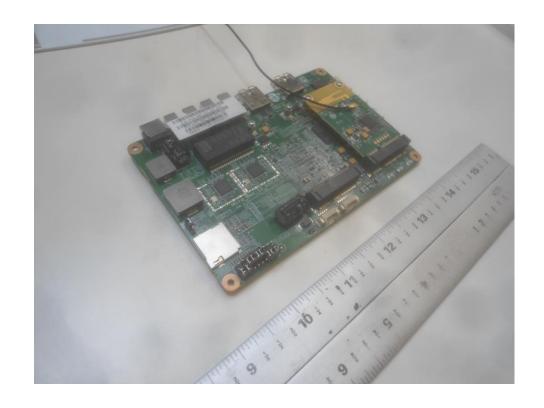
### Flight Test Configuration

- ShipsLog coordinates time stamps of disparate devices and stores data frames.
- Each device is polled for single or blocks of sample intervals as appropriate.
- Device agnostic layer allows us to treat all devices as DAQ/Card/Channel, even a Serial NMEA data source.
- NMT needs combination of background environment data as well as the structural signals.
- IDI needs environmental background and proof of concept data.



### **IDI Basic Compute Module**

- Prototype unit shown is for ground equipment but with some modifications is good for short term use in LEO.
- Our own compute module for longer term space flight (longer than safe with COTS), will come when we have investment or sufficient internal revenue generation to support it.
- Will have just what is needed for our requirements and no more.
- Module will be common across all of our products (microDAQ, ShipsLog, ShipsStore,etc). These vary at the software and daughter board level.



Standard compute module prototype

### **Data Acquisition Unit**



8 port analog microDAQ prototype

- Prototype unit shown is for ground equipment but with some modifications is good for short term use in LEO.
- We are planning our own compute module for long term space flight based on lessons learned.
- Each top hat (1 shown) handles 8
   analog inputs; up to 8 are allowed, giving

   64 inputs.
- Far cheaper, lower volume and lower mass than Lynx DAQ, although it could handle 124 inputs.

### Standardized Sensor Inputs

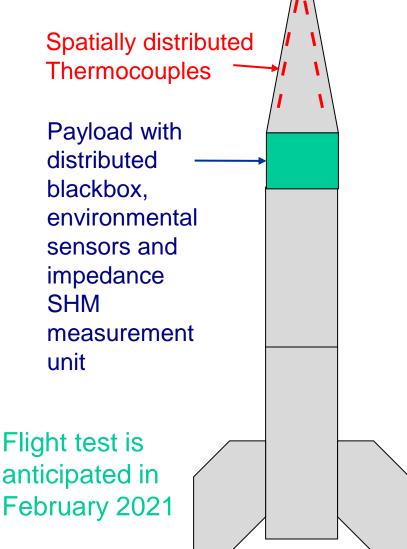
- System supplies power to sensors that require it.
- If sensors require specialized electrical interfaces, we isolate them via instrumentation amplifiers.
- Software loads specialized type objects as needed for conversion of a sensor voltage input to a calibrated engineering unit value.
- Sensors considered:
  - ADXL356 accelerometer 3 axis/ 20g;
  - P51-15-A-B-P pressure transducer;
  - Duplex Insulated K-type thermocouples;
  - 9 additional channels (strain/temperature)



3 Input thermocouple instrumentation amplifier / signal converter

# Preparation for Suborbital Test Flight

- Flight will demonstrate
  - Functionality of distributed blackbox in space and reentry environments
  - Operation of NMT's single channel impedance measurement SHM unit
  - Collection and storage of spatially distributed temperature data.



#### Publications, Presentations, Awards, & Recognitions

#### **PUBLICATIONS**

- Zagrai, A., Campisi, M., Anderson, M., Hunter, D., Sanchez, J.A., Demidovich, N., Kessler, S., (2019) "Structural Diagnostics, Prognostics and Health Management for Future Space Vehicles: Development, Implementation and Testing," Proceedings of 2019 IEEE Aerospace Conference, Big Sky, MT, USA, March 2-9, 2019, pp. 1-11. doi: 10.1109/AERO.2019.8741659
- 2. Zagrai, A., Misla, A., Sanchez, J., Powell, D (2020) "Electro-Mechanical Impedance Method for Structural Health Monitoring of Space Structures: from Laboratory Experiments to Measurements during Spaceflight," Proceedings of the AIAA Propulsion and Energy 2020 Forum, paper AIAA 2020-3529, August 24-28, 2020, Virtual Event, https://doi.org/10.2514/6.2020-3529.
- 3. Amon, D., (2020) "Design of an Inexpensive Black Box for Commercial Orbital and Suborbital Vehicles," presentation at Commercial and Government Responsive Access to Space Technology Exchange (CRASTE), June 22, 2020, Virtual Conference.

#### **PRESENTATIONS**

1. Zagrai, A., Powell, D., Hunter, D., Anderson, M., Amon, D., Demidovich, D., (2019) "Advances in Electro-Mechanical Impedance Structural Health Monitoring for Space Systems," presentation at Commercial and Government Responsive Access to Space Technology Exchange (CRASTE), June 24 - 27, 2019, Henderson, NV.

### **Conclusions and Future Work**

- NMT team prepares a single-channel impedance measurement unit for suborbital test flight
- Establishing of specifications and practical realization of spacecraft's distributed data acquisition system is in progress
- Immortal Data did much work on hardware and software for the proposed system. Due to pandemic, NMTs experiences delays in multi-channel unit and software integration.
- Covid19 impacts operation, especially student hiring and work
- Immortal data and NMT are preparing for a suborbital flight test early next year, likely with UPAerospace.